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A COMPUTER PROGRAM FOR HYDROSTATIC BEARINGS INCLUDING  
THE EFFECTS OF NON-UNIFORM FILM THICKNESS AND RELATIVE  
VELOCITY FOR VARIOUS METHODS OF LUBRICANT SUPPLY

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by

J. G. Hinkle  
V. Castelli  
H. C. Rippel  
C. D. Zimmerman, Jr.

April, 1964

Prepared for

CALIFORNIA INSTITUTE OF TECHNOLOGY  
Jet Propulsion Laboratory

Purchase Order No. BP3 - 211570

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**THE FRANKLIN INSTITUTE**  
LABORATORIES FOR RESEARCH AND DEVELOPMENT  
PHILADELPHIA

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ABSTRACT

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This is a final report describing the development and use of a computer program for the determination of the load carrying capacity, flow requirements, and righting moments of hydrostatic bearings using an incompressible fluid, including the effects of variable film thickness, relative velocity, and method of lubricant supply. The basic equations, numerical approximations, method of solution, numerical treatment and Fortran Program are presented along with instructions on the use of the program and a sample problem. This work is an extension of that reported in Final Report No. F-B2015 dated January 11, 1963, prepared for California Institute of Technology, Jet Propulsion Laboratory.

*[Signature]*

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## INTRODUCTION

The increasing demands for bearings to support heavy loads have caused the development of hydrostatic lubrication to proceed at an accelerated pace. This was accompanied by a need for more sophisticated analysis of bearing characteristics. For some years hydrostatic bearings have been analyzed by techniques that fell short of a rigorous analysis. This was because of the impossibility of producing by hand an exact analytical solution and, therefore, the necessity for resorting to techniques yielding approximate results, e.g., the electric analog.<sup>1\*</sup>

One of the major shortcomings of the conducting-sheet electric-analog is its inability to treat problems involving non-uniform film thicknesses and relative velocity of bearing members. These problems can often be adequately treated by numerical integration techniques, the utilization of which is only hampered by the extremely lengthy and tedious calculations they require. However, the recent development of high speed digital computers has eliminated this drawback, and numerical integration techniques are at present quite feasible.

In the case at hand, hydrostatic bearings are to be designed for the support of a large radio telescope antenna. A number of rectangular pads will transfer the load of the moving superstructure through the pressurized film of oil to a stationary member on the concrete base. While at first glance the analysis of the bearing (flat slider) appears to offer no problem; in reality, the unit loads are high enough to give rise to non-negligible distortions of the bearing members thus requiring the treatment of variable film thicknesses. While, the particular application involves relatively slow velocity of bearing members, there will be a significant contribution of hydrodynamic effects particularly for favorable film clearance distributions. A previous work<sup>2</sup> describes in detail the development of a computer program which did not include the effect of relative velocity of bearing members.

\*Superscript numerals refer to references at the end of this report.

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This report is a complete development of the computer program for the evaluation of pressure distributions, loads, moments, and flows in a hydrostatically-lubricated, rectangular bearing-pad with four or six symmetrically placed rectangular recesses and includes the effect of velocity. The lubricant is to be incompressible, the clearance uniform or non-uniform, and the fluid supply one of the three specified types:

1. Separate pumps feeding each recess
2. Separate pumps feeding opposite pairs of recesses with capillary compensation.
3. Common manifold feeding all recesses with capillary compensation.

This report contains details of the hydrodynamic equations used to analyze the problem, of the numerical technique used for solution, and of the Fortran Program executing the solution complete with block diagrams and flow charts. A guide chart is furnished complete with all the information necessary for using the program and compiling the appropriate input data cards. A sample problem complete with output is also furnished.

## BASIC EQUATIONS AND NUMERICAL APPROXIMATIONS

Utilizing the usual lubrication approximations for a continuous film bearing operating with an incompressible lubricant, Reynolds Equation governs the pressure distribution in the clearance space. For bearings with relative motion, Reynolds Equation assumes the following form<sup>3</sup>

$$\frac{\partial}{\partial x} \left( h^3 \frac{\partial p}{\partial x} \right) + \frac{\partial}{\partial y} \left( h^3 \frac{\partial p}{\partial y} \right) = - 6\mu U \left( \frac{\partial h}{\partial x} \right) \quad [1]$$

For a bearing of rectangular geometry the length of one of the sides can be used as a characteristic length (say L). Then the following dimensionless quantities can be defined in an x-y rectangular coordinate system:

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$$X = \frac{x}{L},$$

$$Y = \frac{y}{L},$$

$$H = \frac{h}{c},$$

$$P = \frac{p - p_a}{p_{ref} - p_a}$$

where  $c$  is a characteristic film thickness (any value),  $p_{ref}$  is a reference pressure (any value),  $p_a$  is the ambient pressure (any value), and  $h$  and  $p$  refer to film thickness and pressure respectively at location  $(x,y)$ .

Reynolds equation now assumes the dimensionless form

$$\left( \frac{\partial^2 P}{\partial X^2} + \frac{\partial^2 P}{\partial Y^2} \right) + \frac{3}{H} \left( \frac{\partial H}{\partial X} \frac{\partial P}{\partial X} + \frac{\partial H}{\partial Y} \frac{\partial P}{\partial Y} \right) = - \frac{\Lambda}{H^3} \frac{\partial H}{\partial X} \quad [2]$$

$$\text{where } \Lambda = \frac{6\mu UL}{c^2 (p_{ref} - p_a)}$$

Three types of bearing feed will be used:

- (a) Specified flow to each recess as given by a positive displacement pump feeding the recess.
- (b) Specified total flow to pairs of diametrically opposite recesses as fed by a single constant displacement pump through capillary compensation.
- (c) Single manifold having a specified supply pressure feeding all recesses through capillary compensation.

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METHOD OF SOLUTION

The solution of equation [2] can be split into two parts;

- a) a particular solution of the non-homogenous equation.
- b) the solution of the homogeneous equation.

Part (a) is accomplished by solving equation [2] numerically with the conditions that  $P = 0$  on the boundaries and all recess areas. This solution shall be called  $P_A(X, Y)$ . Due to the linearity in  $P$  of equation [2], part (b) can be accomplished by the method of superposition. Namely, component solutions,  $P_j(X, Y)$ , are obtained corresponding to a value of the dimensionless pressure,  $P$ , equal to 1 in the  $j^{\text{th}}$  recess and 0 in all other recesses and at the free boundary. Then the pressure distribution,  $P(X, Y)$ , corresponding to any operating condition is expressible as the linear combination,

$$P(X, Y) = \sum_j \alpha_j P_j(X, Y) + P_A(X, Y) \quad [3]$$

where  $\alpha_j$  is a dimensionless number representing pressure in the  $j^{\text{th}}$  recess. Defining;

$Q_{ij}$  = flow out of  $i^{\text{th}}$  recess corresponding to  $j^{\text{th}}$  component solution.

$Q_{iA}$  = flow out of  $i^{\text{th}}$  recess corresponding to  $P_A(X, Y)$  solution.

The dimensionless flow  $q_i$ , out of the  $i^{\text{th}}$  recess when the pressure distribution  $P(X, Y)$  exists is,

$$q_i = \sum_j \alpha_j Q_{ij} + Q_{iA} \quad [4]$$

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Integration of the pressure distribution and evaluation of the moments of the pressure distribution with respect to two non-parallel axes (say X and Y) produces the total load (W) and the location of the center of pressure ( $\xi$ ,  $\eta$ ) in the following manner (see Nomenclature):

$$W_j = \iint P_j(X, Y) dX dY \quad j=1, \dots, N, \Lambda \quad [5]$$

$$\xi_j = \iint \frac{P_j(X, Y) X dX dY}{W_j} \quad j=1, \dots, N, \Lambda \quad [6]$$

$$\eta_j = \iint \frac{P_j(X, Y) Y dX dY}{W_j} \quad j=1, \dots, N, \Lambda \quad [7]$$

For the total solution:

$$W = \sum_j \alpha_j W_j + W_\Lambda , \quad [8]$$

$$\xi = \sum_j \frac{\alpha_j \xi_j W_j}{W} + \frac{\xi_\Lambda W_\Lambda}{W} , \quad [9]$$

$$\eta = \sum_j \frac{\alpha_j \eta_j W_j}{W} + \frac{\eta_\Lambda W_\Lambda}{W} , \quad [10]$$

The attractive feature of the method of superposition is that a given set of component solutions can be utilized with any number of feeding methods as long as the bearing geometry is not altered. The computer program is equipped to take advantage of this feature. Specification of the values of the pertinent parameters in the feeding equations leads to the determination of the appropriate values of the coefficients  $\alpha_j$ .

In correspondence to the three feeding methods under consideration the determining equations are:

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A. Specified Flow to Each Recess:

It is necessary to solve the system

$$q_i = \sum_{j=1}^N \alpha_j Q_{ij} + Q_{iA} \quad (i = 1, \dots, N) \quad [11]$$

where

$q_i$  = specified quantities (dimensionless)

$Q_{ij}, Q_{iA}$  = known quantities from component solutions (dimensionless)

Equations (11) form a non-homogeneous system of  $N$  algebraic equations in  $N$  unknowns (where  $N$  is the number of recesses) which can be solved by conventional methods.

B. Specified Flow to Pairs of Recesses With Capillary Compensation.

The following equations hold:

$$QQQ^{(1)} = q_1 + q_2 \quad [12]$$

$$QQQ^{(2)} = q_3 + q_4 \quad [13]$$

$$QQQ^{(3)} = q_5 + q_6 \quad \left. \begin{array}{l} \\ \end{array} \right\} \text{only for } N = 6 \quad [14]$$

$$q_1 = f_1(p^{(1)} - \alpha_1) \quad [15]$$

$$q_2 = f_2(p^{(1)} - \alpha_2) \quad [16]$$

$$q_3 = f_3(p^{(2)} - \alpha_3) \quad [17]$$

$$q_4 = f_4(p^{(2)} - \alpha_4) \quad [18]$$

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$$\left. \begin{array}{l} q_5 = f_5(p^{(3)} - \alpha_5) \\ q_6 = f_6(p^{(3)} - \alpha_6) \end{array} \right\} \text{only for } N = 6 \quad [19]$$

$$q_i = \sum_{j=1}^N \alpha_j Q_{ij} + Q_{iA} \quad (i = 1, \dots, N) \quad [20]$$

$$q_i = \sum_{j=1}^N \alpha_j Q_{ij} + Q_{iA} \quad (i = 1, \dots, N) \quad [21]$$

The system of  $2.5N$  equations [12] through [21] has  $2.5N$  unknowns  $q_i$ ,  $\alpha_{j=i}$  ( $i = 1 \dots N$ ),  $p^{(K)}$  ( $K = 1 \dots N/2$ ) and can be solved by conventional methods. The quantities  $Q_{ij}^{(K)}$  ( $K = 1 \dots N/2$ ) are specified and the  $f_i$ 's are known characteristics of the  $N$  capillaries.

C. Common Reservoir With Specified Pressure and Capillary Feed to Each Recess.

The following equations hold

$$q_i = f_i(p^{(f)} - \alpha_i) \quad i = 1, \dots, N \quad [22]$$

$$q_i = \sum_{j=1}^N \alpha_j Q_{ij} + Q_{iA} \quad i = 1, \dots, N \quad [23]$$

This system of  $2N$  equations in the  $2N$  unknowns  $q_i$  and  $\alpha_i$  ( $i = 1, \dots, N$ ), can be easily solved by conventional methods.

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NUMERICAL TREATMENT

Equation [2] is approximated by numerical methods with the adoption of "three point, central difference" formulae. Thus the bearing area is divided in K by M rectangular elements, in the x and y directions respectively. The pressure distribution is represented by these values at the nodal points of the resulting grid.

The coordinates of the nodal points are

$$x_k = (i - 1) \Delta X, \quad [24]$$

$$y_m = (j - 1) \Delta Y, \quad [25]$$

where i and j designate the location of a particular nodal point in the X and Y directions respectively. (Note that this use of i and j differs from the use of i and j as recess designations.)

For the pressures

$$p_{k,m} = p(x_k, y_m), \quad [26]$$

$$\frac{\partial p}{\partial x} \Big|_{x_k y_m} = \frac{p_{i+1,j} - p_{i-1,j}}{2 \Delta X}, \quad [27]$$

$$\frac{\partial p}{\partial y} \Big|_{x_k y_m} = \frac{p_{i,j+1} - p_{i,j-1}}{2 \Delta Y}, \quad [28]$$

$$\frac{\partial^2 p}{\partial x^2} \Big|_{x_k y_m} = \frac{p_{i+1,j} - 2p_{i,j} + p_{i-1,j}}{\Delta X^2} \quad [29]$$

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$$\left. \frac{\partial^2 P}{\partial Y^2} \right|_{X_k Y_m} = \frac{P_{i, j+1} - 2P_{i, j} + P_{i, j-1}}{\Delta Y^2}, \quad [30]$$

Equation [2] becomes

$$\begin{aligned} & \frac{P_{i+1, j} + P_{i-1, j}}{\Delta X^2} + \frac{P_{i, j+1} + P_{i, j-1}}{\Delta Y^2} - 2P_{i, j} \left( \frac{1}{\Delta X^2} + \frac{1}{\Delta Y^2} \right) \\ & + \frac{3}{H_{k,m}} \left( \frac{P_{i+1, j} - P_{i-1, j}}{2\Delta X} \left. \frac{\partial H}{\partial X} \right|_{X_k Y_m} + \frac{P_{i, j+1} - P_{i, j-1}}{2\Delta Y} \left. \frac{\partial H}{\partial Y} \right|_{X_k Y_m} \right) \\ & = - \frac{\Lambda}{H^3} \frac{\partial H}{\partial X} \end{aligned} \quad [31]$$

which is solved for  $P_{k,m}$ .

$$P_{k,m} \text{ (evaluated)} = \frac{1}{2(1+\alpha)} \left[ \begin{array}{l} P_{i+1, j} + P_{i-1, j} + \left\{ \begin{array}{l} P_{i+1, j} - P_{i-1, j} \\ P_{i, j+1} + P_{i, j-1} + (P_{i, j+1} - P_{i, j-1})(H_{i,j}) \end{array} \right\} \\ + \frac{2\Delta X}{3H_{k,m}^2} \end{array} \right] (HX_{i,j}) + \alpha \left[ \begin{array}{l} P_{i, j+1} + P_{i, j-1} + (P_{i, j+1} - P_{i, j-1})(HY_{i,j}) \\ \left[ \frac{\Delta X}{\Delta Y} \right]^2 \end{array} \right]$$

where

$$HX_{i,j} = 1.5 \left. \frac{\partial H}{\partial X} \right|_{i,j} \frac{\Delta X}{H_{i,j}}$$

$$HY_{i,j} = 1.5 \left. \frac{\partial H}{\partial Y} \right|_{i,j} \frac{\Delta Y}{H_{i,j}}$$

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Starting with an assumed pressure distribution ( $P = 1$  in the  $j^{th}$  recess and  $P = 0$  at the boundaries and in all other recesses for the  $j^{th}$  component solution or  $P = 0$  at the boundaries and in all recesses for the  $N + 1$  component solution), a value of the pressure,  $P'_{k,m}$ , at a grid point is evaluated in terms of the pressure at its four immediately neighboring grid points.

Assuming

$$P'_{k,m} = \gamma P_{k,m}^{(\text{evaluated})} + (1 - \gamma) P_{k,m}^{(\text{old})} \quad [32]$$

to be a new pressure distribution, the process is repeated until negligible changes are obtained from any one iteration.  $\gamma$  is called a "relaxation factor" and its magnitude sets the rate of growth of the pressure distribution. Limiting the value of  $\gamma$  is the well known phenomena of numerical instability.

For excessive values of  $\gamma$ , the rate of growth can be seen to increase steadily in time thereby indicating lack of convergence of the iteration. The occurrence of this phenomenon is internally detected and is automatically eliminated by successive reduction of the value of  $\gamma$  by the factor 0.8. An initial value of  $\gamma = 1.5$  was used and was revealed to be overly-optimistic. Hence, an initial value of 1.1 was adopted for this program in order to avoid waste of computer time in unstable interations.

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COMPUTER PROGRAM

The computer program is functionally separated into the following four sub-programs.

(1) Subroutine FORMH

Generation of the bearing geometry gridwork and the clearance distribution either by evaluation of appropriate analytic function or by direct input.

(2) Subroutine TILTH

Tilting by any specifiable amount of the clearance distribution already existing in core storage.

(3) Subroutine REYN

Generation of the component solutions  $P_j$  ( $X, Y$ ) and the corresponding component solution values of the loads  $W_j$ , center of pressure coordinates  $\xi_j$ , and  $\eta_j$ , and flows  $Q_{i,j}$ .

(4) Subroutine FLOW

Matching of the component solutions with the proper feeding equations and evaluation of the corresponding pressure distribution  $P$  ( $X, Y$ ), load  $W$ , center of pressure coordinates  $\xi$  and  $\eta$ , and flows out of each recess,  $q_j$ , for any specified feeding conditions.

The program is such that for any given clearance distribution and velocity, the loading and moment results for any of the other feeding methods can be evaluated by repeated, direct entry into Subroutine Flow. It is also possible to make available to the computer the essential results of component solutions  $W_j$ ,  $\xi_j$ ,  $\eta_j$ ,  $Q_{i,j}$  by direct input, so that new feeding conditions can be studied in combination with component solutions obtained during a previous group of runs.

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The internal generation of the clearance distribution is executed by a function of the following type.

$$\text{Define } X - X_o = s \quad \text{where} \quad X_o = A_{22}$$

$$Y - Y_o = t \quad Y_o = A_{23}$$

then

$$\begin{aligned}
 H_1(X, Y) = & A_1 + A_2 s + A_3 t + A_4 s^2 + A_5 t^2 + A_6 st + A_7 s^3 \\
 & + A_8 t^3 + A_9 s^2 t + A_{10} st^2 + A_{11} \sqrt{A_{12} + A_{13} s^2 + A_{14} t^2} \\
 & + A_{15} \cos(A_{16}s) + A_{17} \cos(A_{18}t) + A_{19} \cos(A_{16}s) \cos(A_{18}t) \\
 & - A_{20} \left\{ e^{-A_{21}X} \cos(A_{21}X) + e^{-A_{21}(1-X)} \cos[A_{21}(1-X)] \right. \\
 & \left. - 2e^{-\frac{1}{2}A_{21}} \left[ \cos(\frac{1}{2}A_{21}) \right] \right\} \quad [33]
 \end{aligned}$$

The inclusion of the part of the preceding expression containing the coefficients  $A_{20}$  and  $A_{21}$  was motivated by the results of analyses of the deformation of a beam on elastic foundations under a uniformly distributed load over a finite region<sup>4</sup>.

The constants  $A_{20}$  and  $A_{21}$  are defined as,

$$A_{20} = \frac{\text{load per unit length}}{2kc}$$

$$A_{21} = \Lambda' = \lambda L$$

where

$$\lambda = \sqrt[4]{k/4 E I}$$

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It should be noticed that these terms detract from any clearance distribution given by the remainder of the formula an amount equal to the difference between the deflection at the center and the deflection at any point. It is also the task of FORMH to compile an internal tabulation of the derivatives of the clearance distribution necessary for the solution.

Subroutine TILTH acts by the use of the following formula:

$$H(X,Y) = H_1(X,Y) + T_x(X-X_1) + T_y(Y-Y_1) \quad [34]$$

where:  $H_1(X,Y)$  represents the distribution of clearance which is to be modified (tilted).

This corresponds to saying that the line,

$$T_x(X - X_1) + T_y(Y - Y_1) = 0$$

is the hinge and that the bearing is tilted in space by an amount,

$$\theta = T_x^2 + T_y^2 \quad [35]$$

This feature is particularly useful in the evaluation of restoring moments corresponding to misalignments. The use of TILTH is optional.

In the use of subroutine REYN, it is important to adopt an appropriate value of the truncation constant (TRUNC). Indeed, a large value of TRUNC will accept component solutions that are only a rough approximation to the asymptotic solution, whereas, exceedingly small values of TRUNC will result in wastefully long computation time.

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Originally, TRUNC was evaluated within the computer program as the change in the summation of pressure values at all of the active sill points from one iteration to the next divided by the number of active sill points. This was found to be unrealistic in that for a given value of TRUNC the degree of convergence varied considerably depending upon total grid size ( $X + 1$  by  $M + 1$ ) and relative recess sizes (with respect to bearing size). After further study it was found that a more realistic truncation criteria is the change in the summation of pressure values at all of the active sill points from one iteration to the next multiplied by the total number of grid points and divided by the product of  $\gamma$  and the summation of the pressure values at all of the active sill points.

Figure 1 shows the number of iterations required (as a function of TRUNC) to satisfy various assigned values of TRUNC for a typical large-size grid. Figure 2 shows the convergence of the dimensionless recess pressures and dimensionless recess loads. It can be seen that a truncation constant (TRUNC) of approximately 0.3 yields results that agree with the probable asymptotic value within less than one percent and require approximately 230 iterations. This is a reasonable compromise between accuracy and computer time. Several other gridwork sizes and configurations investigated have yielded satisfactory results using 0.3 for the truncation constant.

It was mentioned previously that the value of the relaxation factor  $\gamma$  is internally adjusted to cope with numerical instability. However, if persistent occurrence of this phenomenon forces the adoption of a value of  $\gamma$  lower than 0.16, subroutine REYN abandons the solution, and a note to this effect is introduced in the output tape.

Use of subroutine REYN requires the specification of the quantity LITER. Termination of the iteration is forced whenever a number of iterations equal to LITER have been performed regardless of the truncation criterion. As presently set up, the program will allow a

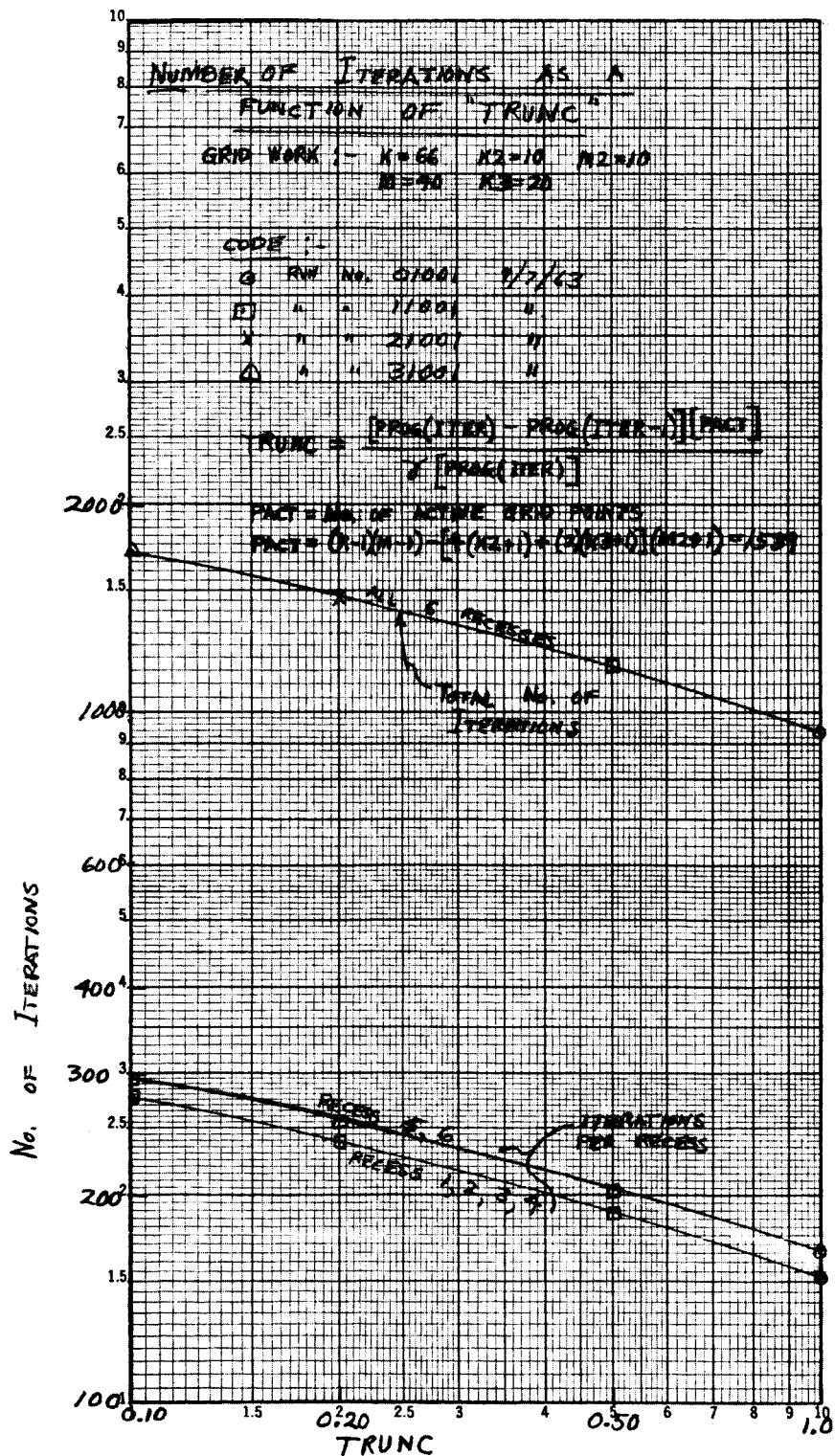


Fig. 1 - NUMBER OF ITERATIONS AS A FUNCTION OF "TRUNC"

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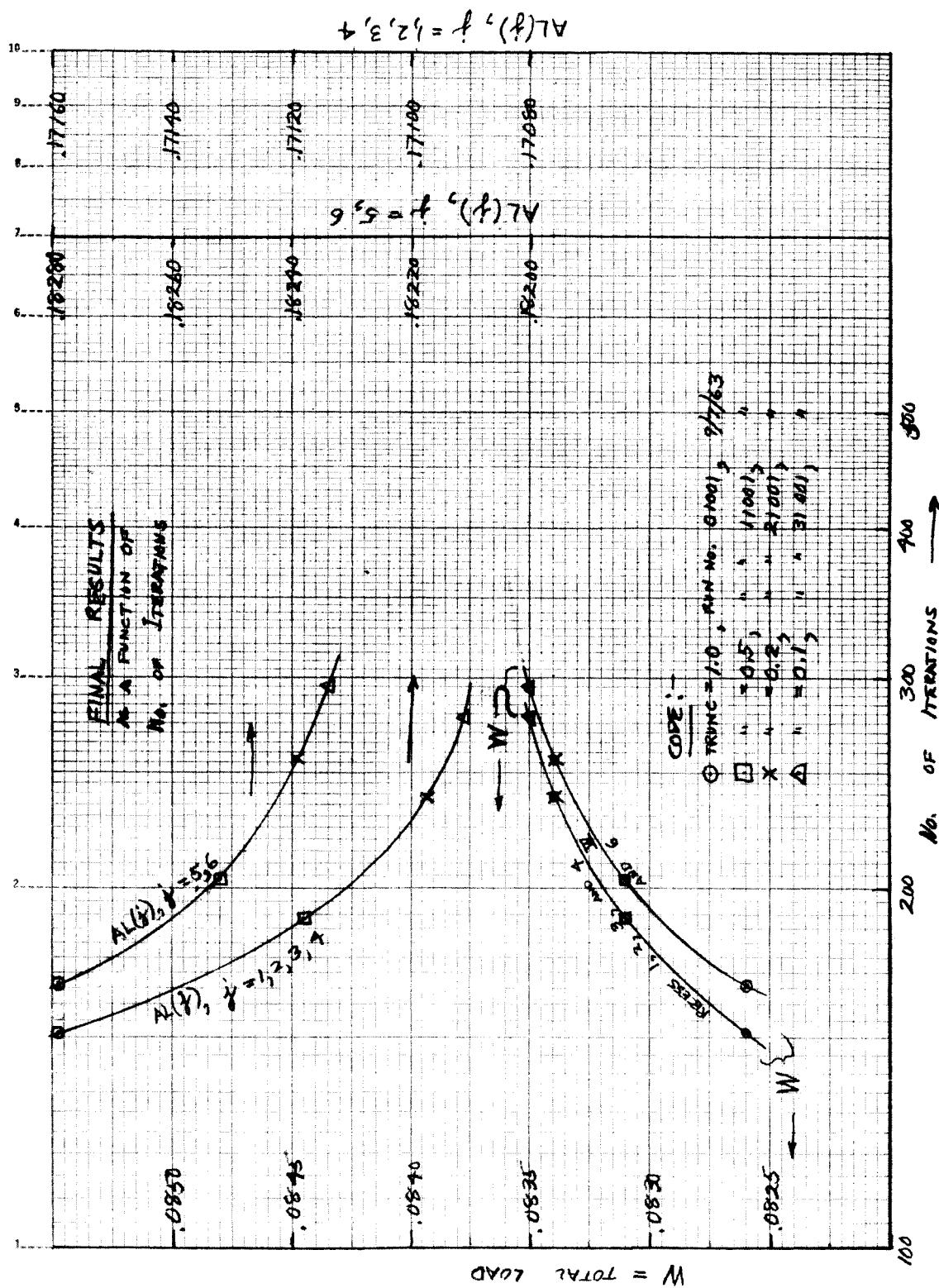


Fig. 2 - FINAL RESULTS AS A FUNCTION OF THE NUMBER OF ITERATIONS

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value of LITER up to 1000. If larger values become desirable, the dimension of PROG in Subroutine REYN should be altered to the maximum value of LITER.

The operation of Subroutine FLOW is controlled by the specification of NCASE which, assuming one of the values 1, 2, or 3, indicates the adoption of the first, second, or third feeding method respectively.

The quantity IOUT enables the user to obtain the pressure and clearance profiles on the output tape ( $IOUT = 1$ ). A value of  $IOUT = 0$  specifies that output of those distributions is not desired. If  $IOUT = 2$ , the pressure profile only will appear in the output. If  $IOUT = 3$ , the clearance profile only will appear in the output.

The information in Table I will be useful in handling the quantities used in the program.

Subroutine FLOW makes use of a FUNCTION subprogram (FUNCTION DETER) in order to evaluate  $4 \times 4$  or  $6 \times 6$  determinants.

The utilization of the above mentioned sub-programs in the solution of specific problems is coordinated by the MAIN program. Specification of the quantity NSWICH instructs the program to solve one of six possible problems:

NSWICH = 1; CALL EXIT

NSWICH = 2; Solve non-velocity flow problem only with read-in values of  $W_j$ ,  $\xi_j$ ,  $\eta_j$ ,  $Q_{i,j}$ .

NSWICH = 3; Solve new flow problem with component solutions already existing in core storage.

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Table I - PROGRAM QUANTITIES

<u>Equation</u>	<u>Fortran Symbol</u>	<u>Output Name</u>
$w_j = \frac{w_j}{L^2(p_{ref} - p_a)}$	W(I)	W(I)
$\xi_j = \frac{x_j}{L}$	CSI(I)	CSI(I)
$\eta_j = \frac{y_j}{L}$	ETA(I)	ETA(I)
$Q_{i,j} = \frac{q_{rc} (12\mu)}{(p_{ref} - p_a)c^3}$	Q(I,J)	Q(I,J)
$q_i = \frac{q_{ri} (12\mu)}{(p_{ref} - p_a)c^3}$	QQ(I)	QQ(I)
$QQQ(K) = \frac{q_{2ri} (12\mu)}{(p_{ref} - p_a)c^3}$	QQQ(K)	Pump Flows
$f_i = 0.2945 \frac{d^4}{lc^3}$	FF(I)	Capillary Factors F(I)
$p^{(f)} = \frac{p_s - p_a}{p_{ref} - p_a}$	PF	Common Constant Pressure
$p_{i,j} = \frac{p_{i,j} - p_a}{p_{ref} - p_a}$	PPPP(I,J)	Pressure Distribution
$\alpha_j = \frac{(p_r)_j - p_a}{p_{ref} - p_a}$	AL(L)	Recess Pressures, AL(j), j=1-N
$W = \frac{w}{L^2(p_{ref} - p_a)}$	WW	Total Load
$\xi = \frac{\int_{x=0}^M w L}{w L}$	CCSI	CSI
$\eta = \frac{\int_{y=0}^M w L}{w L}$	EETA	ETA

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NSWICH = 4; Entirely new problem with or without velocity.

NSWICH = 5; Tilt clearance distribution existing in core storage and solve resulting problem.

NSWICH = 6; Solve the previous problem considering velocity.

In the event NSWICH = 2, the quantity NIMJ should be specified to inform the computer of the number of recesses in the pad under consideration.

Appendix I contains a block diagram of the entire program, the detailed flow charts, Fortran instruction listings, and IBM 7094 compilation records of the six subprograms (MAIN, FLOW, REYN, TILTH, FORMH, DETER).

Appendix II contains the loading record for use with an IBM 7094. This information is essential if the allowable grid size of 67 x 45 is to be extended.

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USE OF THE PROGRAM AND SAMPLE PROBLEMS

A. Input

In order to use the program, the proper input data for the problem to be solved must be put on cards in a prescribed manner. The Input Guide Chart shown in Appendix III indicates the necessary data to be put on cards, its proper format, and the proper sequence of such cards. Figure 3 shows the standard form which is used to facilitate the preparation of punched cards. All of the input data required for solving a single problem is entered in the proper blocks of Figure 3. The number at the upper right-hand corner of each block is for the convenience of the typist to obtain proper word location when right-indexing is used. The following is a detailed explanation of how to specify the numbers to be entered in Figure 3 for proper execution of the particular problem to be solved.

1. Card No. 1 (always required)

The following input data are required in MAIN program

- a. NSWICH: - Any number 1 through 6 may be entered as dictated by the following:

- 1 = Last problem has been solved. Terminate program.
- 2 = Solve flow problem only using basic solution results contained on input cards. It should be noted that the input form, Fig. 3, does not allow for these required cards. See the Input Guide Chart in Appendix III for the proper formats and card sequence.
- 3 = Solve new flow problem with basic solution already in machine (as obtained from previous problem solved).
- 4 = New problem to be solved.

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HYDROSTATIC BEARING PROGRAM INPUT DATA GUIDE (CROSS OUT CARDS NOT USED)									
	{CARD NUMBER}								Page ____ of ____ Pages
①	ARD								
	1	3	5						
	NSWICH	NIMJ	NIGNAL						
② 1 CARD	3	6	9	12	15	18	21	24	27
	K	M	INDEX 1	INDEX 2	K1	K2	K3	M1	M2
			14		28		42		56
									70
③	A(1)	A(2)		A(3)		A(4)		A(5)	
④	A(6)	A(7)		A(8)		A(9)		A(10)	
⑤ 5 CARDS (INCLUDE ALL 5)	A(11)	A(12)		A(13)		A(14)		A(15)	
⑥	A(16)	A(17)		A(18)		A(19)		A(20)	
⑦	A(21)	A(22)		A(23)		A(24)		A(25)	
⑧ 1 CARD	15		30		45		60		
	X1		Y1		TX		<td></td> <td></td>		
	5		21		37				
⑨ 1 CARD	LITER	TRVNC		PLAN					
	5	7	9						
⑩ 1 CARD	NRUN	NCASE	IOUT						
	12	24		36		48		60	
⑪ 1 CARD									72
	QQ(1) QQQ(1)	QQ(2) QQQ(2)	QQ(3) QQQ(3)	QQ(4)	QQ(5)	QQ(6)			
	10								
⑫ 1 CARD	PF								
	12	24		36		48		60	
⑬ 1 CARD	FF(1)	FF(2)	FF(3)	FF(4)	FF(5)	FF(6)			72
	1								
⑭ 1 CARD	NSWICH								

Fig. 3 - HYDROSTATIC BEARING PROGRAM INPUT DATA GUIDE

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5 = The distribution of clearance already in the machine will be tilted and new basic solutions and final results for the new clearance distribution will be obtained.

6 = The previous zero-velocity problem will be solved considering velocity.

- b. NIMJ: - If NSWICH = 2 is used, then the value of NIMJ is equal to the total number of recesses. Otherwise NIMJ is left blank.
  - c. NIGNAL: - A 1 is entered here if velocity (PLAM) is to be considered. Otherwise PLAM will not be used by the program.
2. Card No. 2 (Only required when NSWICH = 4)
- The following input data are required in Subroutine FORMH
- a. K: - This is the number of cells in the X-direction formed by the gridwork which simulates the pad. (See Fig. 5). K must be even. The maximum value of K allowed is 66\*. (See Section B below.)
  - b. M: - This is the number of cells in the Y-direction formed by the gridwork. (See Fig. 5). M must also be even. The maximum value of M allowed is 44\*. (See Section B below.)
  - c. INDEX 1: - If the clearance distribution is to be read in from cards then the value 1 is used for INDEX 1. It should be noted that the standard input form, Fig. 3, does not allow for these required cards. Consult the Input Guide Chart in Appendix III for the proper format and card sequence. If the clearance distribution is to be generated within the machine, INDEX 1 is not equal to 1 (normally left blank).
  - d. INDEX 2: - If the clearance distribution as read in from cards or generated internally (per INDEX 1) is to be tilted, then INDEX 2 = 1. Otherwise, INDEX 2 is not equal to 1.

---

\*The allowable gridwork size  $(K + 1) \times (M + 1)$  may be increased by changing all of the subscripted dimension statements from (67,45) to  $[(K+1)_{\max}, (M+1)_{\max}]$ .

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- e. K1: - This is the number of cells in the X-direction between edges of pad and recesses 1, 2, 3 and 4. (See Section B below.)
  - f. K2: - This is the number of cells representing the length of recesses 1, 2, 3 and 4 in the X-direction. (See Section B below.)
  - g. K3: - This is the number of cells representing the length of recesses 5 and 6 in the X-direction. If only a total of four recesses are to be used, K3 = 0. K3 must be an even number (See Section B below.)
  - h.. M1: - This is the number of cells in the Y-direction between edges of pad and all of the recesses. (See Section B below.)
  - i. M2: - This is the number of cells representing the width of the recesses in the Y-direction (See Section B below.)
  - j. AL: - This is the ratio of the length of the pad in the Y-direction and the length of the pad in the X-direction (See Fig. 4). AL = LY/LX. (See Section B below.)
3. Card Numbers 3 through 7 (Used only if NSWICH = 4 and INDEX ≠ 1)

The following input data are required in Subroutine FORMH.

- a. A(1) thru A(23): - These are the values of the 23 constants used to generate the clearance distribution within the computer. It is important to note that all five cards must be used even if the values contained on any of them are all zeros.

4. Card No. 8 (Used only when; NSWICH = 4 and INDEX 2 = 1, or NSWICH = 5)

The following input data are required in Subroutine TILTH.

- a. Xl: - The value to be used is the dimensionless distance in the X-direction to the line about which it is desired to rotate the pad. Xl can be any value between 0 and 1. If Xl is 0 then the pad will be rotated about the I = 1 gridwork line. If Xl = 1, the pad will be rotated about the I = (K+1) gridwork line. If LX = 0.5, the pad will be rotated about the center line I = K/2 + 1, and so forth.

The value of Xl need not necessarily fall on a gridwork line.

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- b.  $Y_1$ : - The value to be used is the dimensionless distance in the Y-direction to the line about which it is desired to rotate the pad.  $Y_1$  can be any value between 0 and AL. If  $Y_1 = 0$  then the pad will be rotated about the  $J = 1$  gridwork line. If  $Y_1 = AL$ , the pad will be rotated about the  $J = (M+1)$  gridwork line, and so forth. The value of  $Y_1$  need not necessarily fall on a gridwork line.
- c. TX: - The value of TX required to obtain a specific amount of tilt about  $X_1$  may be obtained using the following equation:

$$TX = [H(1, AL/2) - H(0, AL/2)] - [H_1(1, AL/2) - H_1(0, AL/2)] \quad [36]$$

where  $H(1, AL/2)$  = desired value of H at  $X = 1$ ,  $Y = AL/2$

$H(0, AL/2)$  = desired value of H at  $X = 0$ ,  $Y = AL/2$

$H_1(1, AL/2)$  = previous value of H at  $X = 1$ ,  $Y = AL/2$

$H_1(0, AL/2)$  = previous value of H at  $X = 0$ ,  $Y = AL/2$

- d. TY: - The value of TY required to obtain a specific amount of tilt about  $Y_1$  may be obtained by using the following equation:

$$TY = [H(1/2, AL) - H(1/2, 0)] - [H_1(1/2, AL) - H_1(1/2, 0)] \quad [37]$$

where  $H(1/2, AL)$  = desired value of H at  $X = 1/2$ ,  $Y = AL$

$H(1/2, 0)$  = desired value of H at  $X = 1/2$ ,  $Y = 0$

$H_1(1/2, AL)$  = previous value of H at  $X = 1/2$ ,  $Y = AL$

$H_1(1/2, 0)$  = previous value of H at  $X = 1/2$ ,  $Y = 0$

5. Card No. 9 (used only if NSWICH = 4, 5 or 6)

The following input data are required in Subroutine REYN.

- a. LITER: - Maximum number of iterations allowable to obtain a component solution. When this number of iterations is reached, the iteration procedure will stop and rest of problem will be solved using component solutions obtained.

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- b. TRUNC: - This is the value assigned to the truncation constant. Usually, a value of TRUNC = 0.30 yields adequate convergence. However, it is usually good practice to experimentally evaluate a satisfactory value for a particular grid work pattern by varying the assigned value of TRUNC and observing the convergence of the final results.
- c. PLAM: - The number assigned to PLAM is  $\Lambda$ , where

$$\Lambda = \frac{6\mu UL}{(p_{ref} - p_a)c^2}$$

Since  $(p_{ref} - p_a)$  is an unknown value prior to solution, PLAM is usually assigned a magnitude of 1.0 although any number may be used.

6. Card No. 10 (always used)

The following input data are required in Subroutine FLOW

- a. NRUN: - The identifying run number is inserted in this block.
- b. NCASE: - The method of feeding code number is inserted in this block per the following:
  - 1 = separate constant displacement pumps feeding each recess.
  - 2 = one constant displacement pump feeding two opposite recesses, i.e., in Fig. 4, one pump feeds recesses 1 and 2, one pump feeds recesses 3 and 4, and the third pump (if used) feeds recesses 5 and 6 with capillary tubes used between pumps and recesses they feed.
  - 3 = common pressure regulated supply manifold feeding each recess through capillary tubes.
- c. IOUT: - The number used here dictates the output desired with regard to clearance and pressure distributions per the following code.
  - 0 = do not print out clearance and pressure distribution.
  - 1 = print out clearance and pressure distribution.
  - 2 = print out pressure distribution only.
  - 3 = print out clearance distribution only.

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7. Card No. 11 (Used only if NCASE = 1 or 2)

The following input data are required in Subroutine FLOW.

- a. QQ(1), ... QQ(N): - If NCASE = 1, the values of  $q_1 \dots q_N$  are inserted in the first N blocks. N is either 4 recesses or 6 recesses.  $q_i$  is the constant flow to each recess.

- b. QQQ(1), ..., QQ(N/2): - If NCASE = 2, the values of  $q_1 \dots q_{N/2}$  to be used are,

$$QQQ(1) = q_1 + q_2$$

$$QQQ(2) = q_3 + q_4$$

$$QQQ(3) = q_5 + q_6 \text{ (if used)}$$

Note: - If NCASE = 3, this card (number 11) is not used.

8. Card No. 12 (Used only if NCASE = 3)

The following input data is required in Subroutine FLOW.

- a. PF: - The number to be used here is the dimensionless manifold pressure,  $p^{(f)}$  where

$$p^{(f)} = \frac{(p_s - p_a)}{(p_{ref} - p_a)} = \frac{q_i}{f_i} + \alpha_j$$

[38]

Since  $q_i$  and  $\alpha_j$  are unknown, a value of  $p^{(f)} = 1.0$  is usually used although any number may be used.

9. Card No. 13 (Used only if NCASE = 2 or 3)

The following input data is required in Subroutine FLOW.

- a. FF(1), ... FF(N): - Any number may be assigned for FF(I) (the capillary constant,  $f_i$ ). N is either 4 or 6 recesses. Hence 4 or 6 values of FF(I) are required.

10. Card No. 14 (Used only if previous cards are last problem to be solved).

The following input data is required in the MAIN program.

- a. NSWICH: - The number 1 is used to indicate that last problem has been solved.

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B. Selection and Specification of Gridwork Pattern to be Used

As has been previously indicated it is necessary that a proper balance be achieved between maximum and minimum number of grid points to be used in order to achieve reasonably accurate results without an undue amount of computer time. It should also be evident that there may have to be some compromise between recess size and locating dimensions and computer program gridwork. That is, some of the recess dimensions desired to be used may have to be slightly modified in order to fit into a finite grid point pattern.

Figure 4 shows the required geometry of a 6-recess pad. Note that if a four recess pad is to be considered, the dimension  $L_{x3}$  is equal to zero. In order to illustrate how a suitable grid work may be selected and specified, consider the pad of Figure 4 to have the following dimensions:

<u>X-Direction</u>	<u>Y-Direction</u>
$L_x = 32.0"$	$L_y = 30.0"$
$L_{x1} = 3.0"$	$L_{y1} = 4.5"$
$L_{x2} = 5.0"$	$L_{y2} = 6.0"$
$L_{x3} = 8.0"$	$L_{y4} = 9.0"$
$L_{x4} = 4.0"$	

The first step is to select the smallest sill dimension in the x-direction ( $L_{x1}$  or  $L_{x4}$  for 6-recesses,  $L_{x1}$  or  $2L_{x4}$  for 4-recesses) and the y-direction ( $L_{y1}$  or  $L_{y4}$ ). For the above problem we would select,

$$L_{x1} = 3.0" \text{ and } L_{y1} = 4.5"$$

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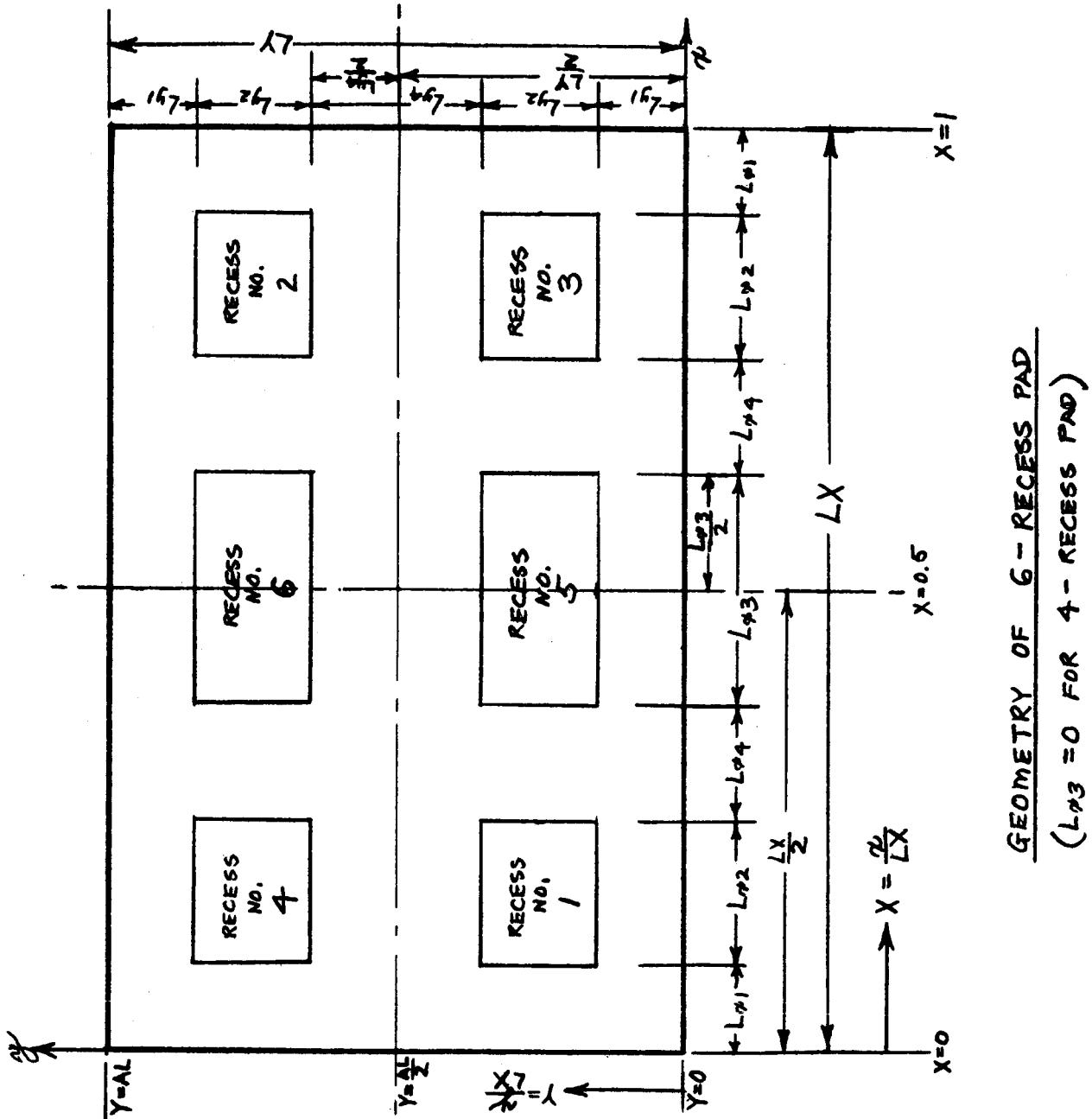


Fig. 4 - GEOMETRY OF 6-RECESS PAD

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We can write the following equations;

$$\left. \begin{array}{l} K_2 = K_1 \left( \frac{L_{x2}}{L_{x1}} \right) \quad M_2 = M_1 \left( \frac{L_{y2}}{L_{y1}} \right) \\ \\ K_3 = K_1 \left( \frac{L_{x3}}{L_{x1}} \right) \quad M_4 = M_1 \left( \frac{L_{y4}}{L_{y1}} \right) \\ \\ K_4 = K_1 \left( \frac{L_{x4}}{L_{x1}} \right) \quad M = M_1 \left( \frac{L_y}{L_{y1}} \right) \\ \\ K = K_1 \left( \frac{L_x}{L_{x1}} \right) \end{array} \right\}$$

[39]

Substituting the pad dimensions into equations [39] yields,

$$\left. \begin{array}{l} K_2 = \frac{5}{3}(K_1) \quad M_2 = \frac{4}{3}(M_1) \\ \\ K_3 = \frac{8}{3}(K_1) \quad M_4 = 2(M_1) \\ \\ K_4 = \frac{4}{3}(K_1) \quad M = \frac{20}{3}(M_1) \\ \\ K = \frac{32}{3}(K_1) \end{array} \right\}$$

[40]

If we now select a number of integer values for  $K_1$  we can evaluate all of the other values of interest. In order to have at least one grid point in the sill,  $K_1$ , and  $M_1$  (or  $K_4$  and  $M_4$  if  $L_{x4} < L_{x1}$ , and  $L_{y4} < L_{y1}$ ) must be at least 2. The following table summarizes the results obtained when various integer values of  $K_1$  are used in equations [40].

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K1	K2	K3 (Even)	K4	K (Even)
2	$3-1/3 = 3$	$5-1/3 = 6$	$2-2/3 = 3$	$21-1/3 = 22$
3	$5 = 5$	$8 = 8$	$4 = 4$	$32 = 32$
4	$6-2/3 = 7$	$10-2/3 = 10$	$5-1/3 = 5$	$42-2/3 = 42$
5	$8-1/3 = 8$	$13-1/3 = 14$	$6-2/3 = 7$	$53-1/3 = 54$
6	$10 = 10$	$16 = 16$	$8 = 8$	$64 = 64$
7	$11-2/3 = 12$	$18-2/3 = 18$	$9-1/3 = 9$	$74-2/3 = 74$
8	$13-1/3 = 13$	$21-1/3 = 22$	$10-2/3 = 11$	$85-1/3 = 86$

M1	M2	M4 (Even)	—	M (Even)
2	$2-2/3 = 3$	4	—	$13-1/3 = 14$
3	$4 = 4$	6	—	$20 = 20$
4	$5-1/3 = 5$	8	—	$26-2/3 = 26$
5	$6-2/3 = 7$	10	—	$33-1/3 = 34$
6	$8 = 8$	12	—	$40 = 40$
7	$9-1/3 = 9$	14	—	$46-2/3 = 46$
8	$10-2/3 = 11$	16	—	$53-1/3 = 54$

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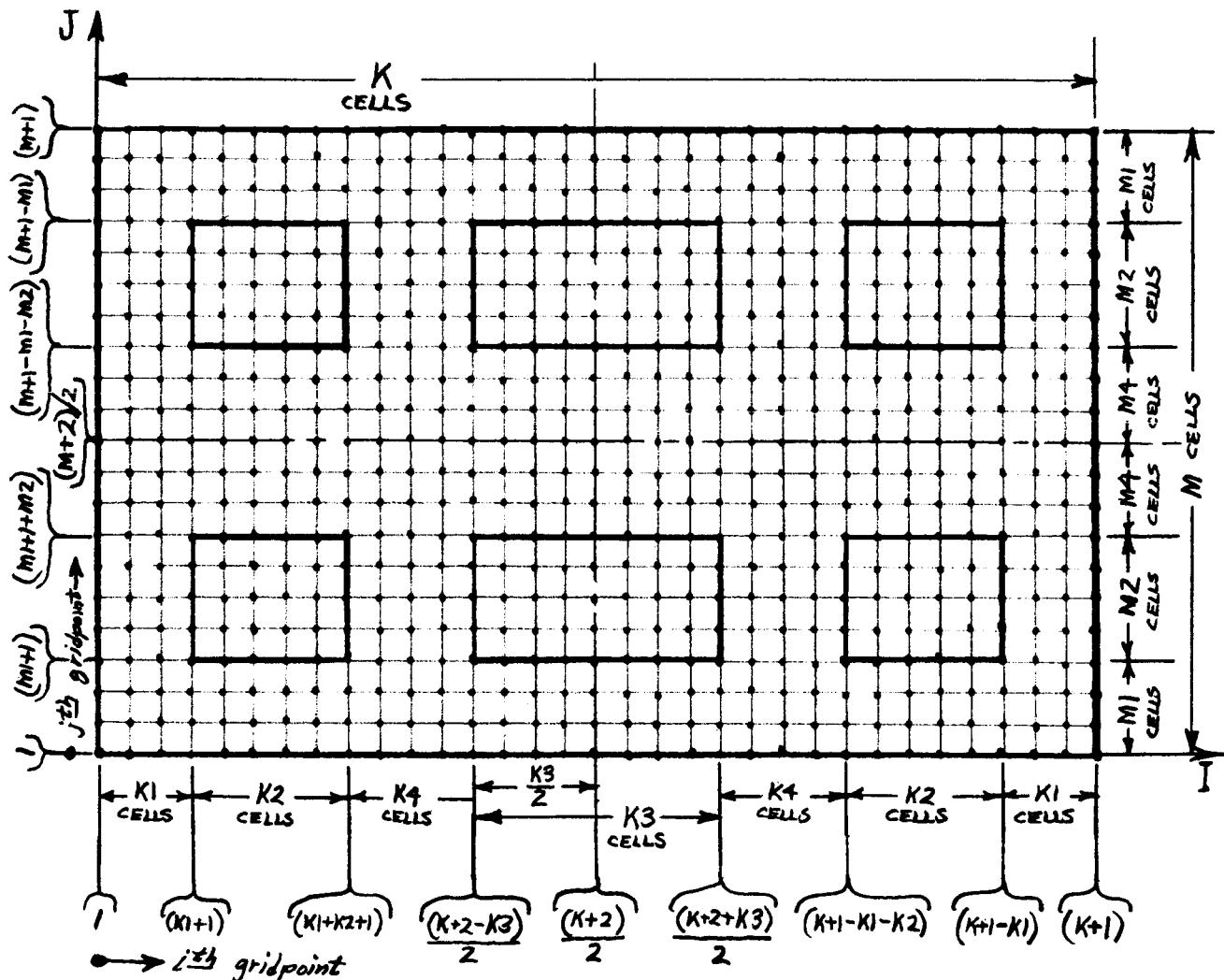
The following items should be noted in the above table.

1. The fractional values when obtained for  $K_2$ ,  $K_4$  are rounded-off to the nearest odd or even integer.
2. The fractional values when obtained for  $K_3$ ,  $K$ ,  $M_4$  and  $M$  are rounded-off to the nearest even integer.
3. The values obtained for  $K_1 = 7$  and  $K_1 = 8$  should not be considered since  $K > 66$ . (maximum value currently permissible).

While it is permissible to use any combination of the above listed values of  $K_1$  and  $M_1$  (except  $K_1 = 7$  or 8) we can further narrow down the combination to be selected by considering the following:

1. The maximum value of  $K_1$ ,  $M_1$ ,  $K_4$  and  $M_4$  should not be greater than 8 in order to yield 7 or less active sill grid points between adjacent recesses or between any recess and the outer periphery of the pad in order to keep computer time within reasonable limits. On this basis we would, therefore, eliminate from consideration  $K_1 = 7$  or 8,  $M_1 = 5, 6, 7$  or 8.
2. While it is permissible to use only one active sill point ( $K_1 = 2$  and  $M_1 = 2$ ) it is not recommended if other permissible values exist. On this basis we would, therefore, eliminate from consideration  $K_1 = 2$  and  $M_1 = 2$ .
3. This leaves us with possible combinations of  $K_1 = 3, 4, 5$  or 6 and  $M_1 = 3$  or 4. Notice that for  $K_1 = 4$  and 5 and for  $M_1 = 4$ , that non-integer values were obtained. This means for these values, the resulting grid work will only approximate the actual pad and recess dimensional relationship. On this basis, we therefore eliminate from consideration  $K_1 = 4$  and 5 and  $M_1 = 4$ .
4. This leaves us with two possible combinations,  $K_1 = 3$ ,  $M_1 = 3$  or  $K_1 = 6$ ,  $M_1 = 3$ . Of these two we would select the smaller grid combination ( $K_1 = 3$ ,  $M_1 = 3$ ) in order to conserve computer time.

The resulting grid work for  $K_1 = 3$  and  $M_1 = 3$  is shown in Figure 5. The values of  $K_1 = 3$ ,  $K_2 = 5$ ,  $K_3 = 8$ ,  $K = 32$ ,  $M_1 = 3$ ,  $M_2 = 4$  and  $M = 20$  are the values that would be entered on input card no. 2 of Fig. 3.



GRIDWORK FOR 6-RECESS PAD

( $K_3 = 0$  FOR 4-RECESS PAD)

Fig. 5 - GRIDWORK FOR A 6-RECESS PAD

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C. Output

The output of the program consists of two result pages and a listing of the pressure and clearance distributions over the grid. The first output page contains:

1. Run number
2. Title containing information on bearing type (hybrid includes velocity), number of recesses, and feeding method.
3. Value of Lambda, if used
4. Grid work data
5. Clearance distribution coefficients (for non-zero values)
6. Values of essential results of the component solutions.
7. Final results
8. Recess pressures

The second page contains:

1. Run number
2. Title containing information on bearing type (hybrid includes velocity), number of recesses, and feeding method.
3. A digit array representing configuration of pad with the following code.  
0 = external boundary point  
1 = recess point  
2 = sill point

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D. Relationships Among the Dimensionless Variables

A number of dimensionless quantities are defined and evaluated by the computer program. These are:

$$\text{Load Factor} = W = \frac{w}{L^2(p_{ref} - p_a)} \quad [41]$$

$$\text{Flow Factor} = q_i = \frac{q_{ri} (l2u)}{c^3(p_{ref} - p_a)} \quad [42]$$

$$\text{Film Thickness Factor} = H_{i,j} = \frac{h_{i,j}}{c} \quad [43]$$

$$\text{Velocity Factor} = \Lambda = \frac{6\mu UL}{c^2(p_{ref} - p_a)} \quad [44]$$

$$\text{Recess Pressure Factor} = \alpha_j = \frac{(p_r)_j - p_a}{(p_{ref} - p_a)} \quad [45]$$

Supply Pressure Factors

Separate pumps feeding each recess (Case 1)

$$\alpha_{js} = \alpha_j \quad [46]$$

One pump feeding two recess thru capillary tubes (Case 2)

$$p_i^{(K)} = \frac{q_i}{f_i} + \alpha_j \quad [47]$$

Each recess fed through capillary tubes from a common supply manifold (Case 3)

$$p^{(f)} = \frac{q_i}{f_i} + \alpha_j = PF \quad [48]$$

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Capillary Factors

$$\text{Case 1; } f_i = \infty \quad [49]$$

$$\text{Case 2; } f_i = \frac{q_i}{(p_i^{(K)} - \alpha_j)} \quad [50]$$

$$\text{Case 3; } f_i = \frac{q_i}{(p_i^{(f)} - \alpha_j)} \quad [51]$$

Moment Factors

Component about X = 0

$$\xi = \frac{M}{wL} \quad [52]$$

Component about Y = 0

$$\eta = \frac{M}{wL} \quad [53]$$

Depending upon the feeding method used, the values of the above mentioned dimensionless quantities are either assigned as input or evaluated within the computer program. The following table indicates whether the quantities are specified as input, (I), appear as output (O), or are able to be evaluated from the output (E).

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Dimensionless Factor	FEEDING METHOD		
	Separate Pumps Feeding Each Recess	Separate Pumps Feeding Two Recesses Through Capillaries	Common Manifold at Fixed Pressure Through Capillaries
W	0	0	0
$q_i$	I	I	0
$H_{i,j}$	I	I	I
$\Lambda$	I	I	I
$\alpha_j$	0	0	0
$\xi$	0	0	0
$\eta$	0	0	0
$p^{(K)}$	-	E	-
$p^{(f)}$	-	-	I
$f_i$	-	I	I

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From the above listed dimensionless factors we can obtain relations useful for evaluating the dimensional quantities of the variables of interest. The following relationships may be derived using the above listed dimensionless factors.

Load-Film Thickness-Flow Coefficient

$$\frac{w(h_{i,j})^3}{L^2 q_{ri} \mu} = \frac{12(w)(H_{i,j})^3}{(q_i)} = (\overline{LHQ}_i) \quad [54]$$

Load-Film Thickness-Velocity Coefficient

$$\frac{L^3 \mu U}{w(h_{i,j})^2} = \frac{(\Lambda)}{6(w)(H_{i,j})^2} = (\overline{LHV}) \quad [55]$$

Film Thickness-Flow-Velocity Coefficient

$$\frac{UL(h_{i,j})}{q_{ri}} = \frac{2(\Lambda)(H_{i,j})}{(q_i)} = (\overline{HQ}_i V) \quad [56]$$

Load-Flow-Velocity Coefficient

$$U \left[ \frac{L^5 \mu}{q_{ri}^2 w} \right]^{1/3} = \frac{2(\Lambda)}{\left[ 12(w)(q_i)^2 \right]^{1/3}} = (\overline{LQ}_i V) \quad [57]$$

Load-Film Thickness-Flow-Velocity Coefficient

$$\frac{w(h_{i,j})^4 U}{q_{ri}^2 L \mu} = \frac{24(w)(H_{i,j})^4 (\Lambda)}{(q_i)^2} = (\overline{LHQ}_i V) \quad [58]$$

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Recess Pressure Coefficient

$$\frac{(p_{rj} - p_a)L^2}{w} = \frac{(\alpha_j)}{(W)} = (\overline{PR_j W}) \quad [59]$$

Supply Pressure Coefficients

Case 1 feed method;

$$\frac{(p_{sj} - p_a)L^2}{w} = \frac{(\alpha_j)}{(W)} = (\overline{PS_j W_1}) \quad [60]$$

Case 2 feed method;

$$\frac{(p_{sj} - p_a)L^2}{w} = \frac{(q_i/f_i) + (\alpha_j)}{(W)} = (\overline{PS_j W_2}) \quad [61]$$

Case 3 feed method;

$$\frac{(p_{sj} - p_a)L^2}{w} = \frac{(q_i/f_i) + (\alpha_j)}{(W)} = (\overline{PS_j W_3}) \quad [62]$$

Pressure Ratio Coefficients

Case I feed method;

$$\frac{(p_{rj} - p_a)}{(p_{sj} - p_a)} = 1.0 = (\overline{\beta_j 1}) \quad [63]$$

Case 2 feed method;

$$\frac{(p_{rj} - p_a)}{(p_{sj} - p_a)} = \frac{(\alpha_j)}{(q_i/f_i) + (\alpha_j)} = (\overline{\beta_j 2}) \quad [64]$$

Case 3 feed method;

$$\frac{(p_{rj} - p_a)}{(p_{sj} - p_a)} = \frac{(\alpha_j)}{(p^{(f)})} = (\overline{\beta_j 3}) \quad [65]$$

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Pad Righting Moment-Load Coefficients (about pad center)

Component (about X = 0.5)

$$\frac{M_x}{wL} = (0.5 - \xi) = (\overline{MXW}) \quad [66]$$

Component [about Y = 0.5 (AL)]

$$\frac{M_y}{wL} = [0.5(AL) - \eta] = (\overline{MYW}) \quad [67]$$

Total moment coefficient (about pad center)

$$\frac{M}{wL} = \left[ (l/2 - \xi)^2 + (AL/2 - \eta)^2 \right]^{\frac{1}{2}} = (\overline{MW}) \quad [68]$$

Direction Angle

$$\phi = \tan^{-1} \left\{ \frac{(AL/2 - \eta)}{(l/2 - \xi)} \right\} = (\overline{MANG}) \quad [69]$$

Pad Righting Moment-Film Thickness-Flow Coefficient

Component about X = 0.5

$$\frac{M_x (h_{i,j})^3}{L^3 q_{ri} \mu} = \frac{12(l/2 - \xi)(w)(h_{i,j})^3}{(q_i)} = (\overline{MXHQ_i}) \quad [70]$$

Component about Y = 0.5(AL)

$$\frac{M_y (h_{i,j})^3}{L^3 q_{ri} \mu} = \frac{12(AL/2 - \eta)(w)(h_{i,j})^3}{(q_i)} = (\overline{MYHQ_i}) \quad [71]$$

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Total moment

$$\frac{M(h_{i,j})^3}{L^3 q_{ri} \mu} = \frac{12(W)(H_{i,j})^3}{(q_i)} \left[ (1/2 - \xi)^2 + (AL/2 - \eta)^2 \right]^{1/2} \quad [72]$$

E. Sample Problems

A flat bearing pad having the dimensions shown in Figure 6 is to be analyzed using the computer program. Each recess is fed from a separate constant displacement pump for the following conditions of clearance distribution and velocity.

<u>Run Number</u>	<u>Clearance Distribution</u>	<u>Velocity</u>
7701	Uniform clearance all over	0
7702	Tilted about $X = 1/2$ , $(H_{1,j} = 0.5), (H_{KK,j} = 1.5)$	0
7703	Tilted about $X = 1/2$ $(H_{1,j} = 0.5), (H_{KK,j} = 1.5)$	$\Lambda = 1.0$

Copies of the input data sheets for these three runs are shown in Appendix IV.

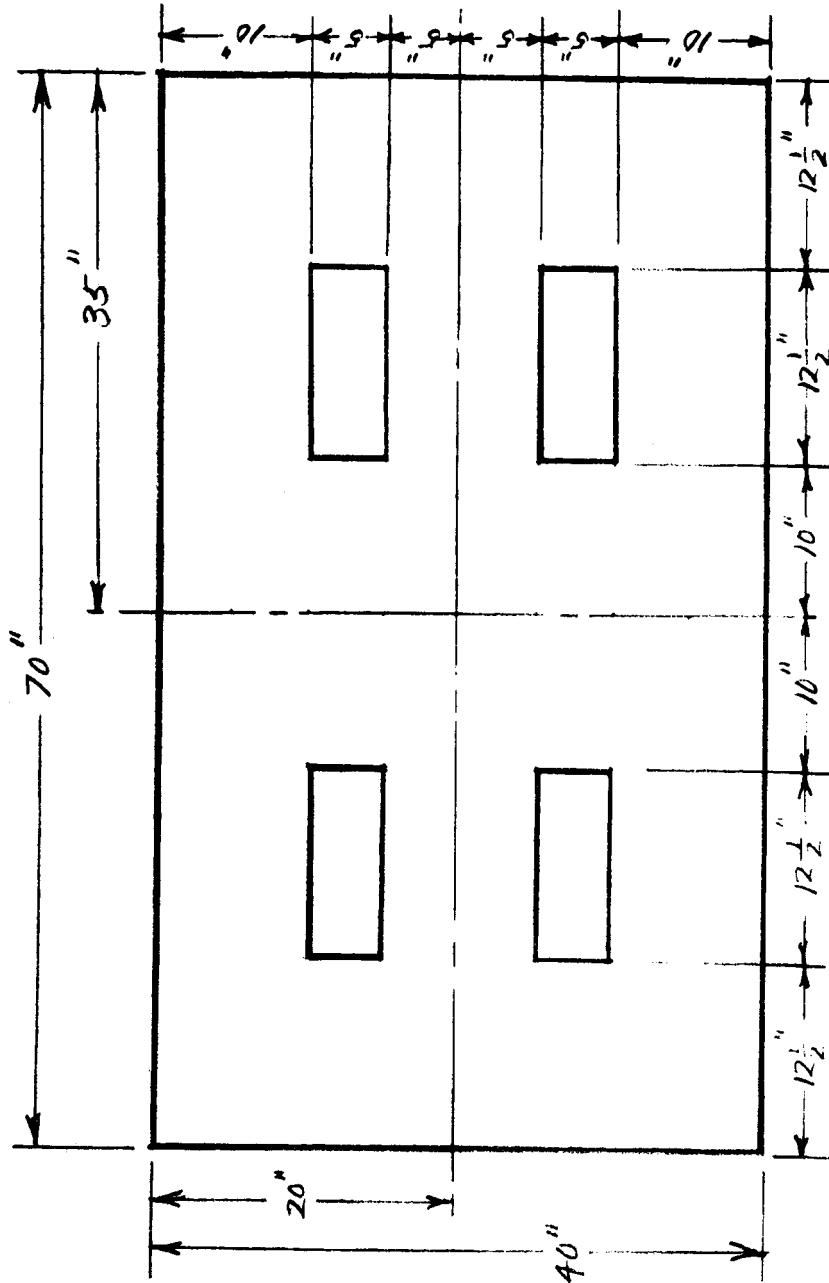
For convenience, let the dimensionless clearance at the center of the pad be

$$H_c = H \left( \frac{K+2}{2}, \frac{M+2}{2} \right) \quad [73]$$

For all three runs,  $H_c = 1.0$

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DIMENSIONS OF PAD FOR

SAMPLE PROBLEM

Fig. 6 - DIMENSIONS OF PAD FOR SAMPLE PROBLEM

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The dimensionless computer input and output data are summarized below.

Dimensionless Factors	Run Number		
	7701	7702	7703
W	0.0933	0.1187	0.1308
$q_i$	1.0	1.0	1.0
$H_c$	1.0	1.0	1.0
$\xi$	0.5000	0.3708	0.3694
$\eta$	0.2857	0.2857	0.2857
$\alpha_1$	0.30184	0.60023	0.64679
$\alpha_2$	0.30184	0.17013	0.18384
$\alpha_3$	0.30184	0.17013	0.18384
$\alpha_4$	0.30184	0.60023	0.64679
$\Lambda$	0	0	1.0

Dimensionless performance coefficients in terms of  $H_c$  may be evaluated using equations [54] through [72], yielding the results listed below:

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Performance Coefficients	Run Number		
	7701	7702	7703
$\overline{LHQ}_i$	1.1196	1.4244	1.5696
$\overline{LHV}$	0	0	1.275
$\overline{HQ}_i V$	0	0	2.0
$\overline{LQ}_i V$	0	0	1.72
$\overline{LHQ}_i V$	0	0	3.14
$\overline{PR}_j W$	3.23	5.07 for 1 & 4 1.44 for 2 & 3	4.95 for 1 & 4 1.40 for 2 & 3
$\overline{PS}_j W_1$	3.23	5.07 for 1 & 4 1.44 for 2 & 3	4.95 for 1 & 4 1.40 for 2 & 3
$\overline{\beta_j}^1$	1.0	1.0	1.0
$\overline{MXW}$	0	0.1292	0.1306
$\overline{MYW}$	0	0	0
$\overline{MW}$	0	0.1292	0.1306
$\overline{MANG}$	0	0	0
$\overline{MXHQ}_i$	0	15.48	15.32
$\overline{MYHQ}_i$	0	0	0
$\overline{MHQ}_i$	0	15.48	15.32

1000 //

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Problem 1: - Determine the recess and supply pressure required for a load  $w = 1.5 \times 10^6$  pounds, and the given pad length of  $L = 70"$ .

The equations to be used are,

$$\frac{(p_{rj} - p_a)L^2}{w} = \overline{PR_j W}$$

$$\frac{(p_{sj} - p_a)L^2}{w} = \overline{PS_j Wl}$$

For the given pad length of  $L = 70"$ .

$$(p_{rj} - p_a) = (\overline{PR_j W}) \frac{1.5 \times 10^6}{4900} = 306(\overline{PR_j W})$$

Since  $\overline{PR_j W} = \overline{PS_j Wl}$ ,

$$(p_{sj} - p_a) = (p_{rj} - p_a) = 306(\overline{PR_j W})$$

Using the computed values of  $\overline{PR_j W}$ , The values of recess pressure and supply pressure (pump discharge) in psig are,

Recess No.	Run Number		
	7701	7702	7703
1	988	1550	1515
2	988	441	428
3	988	441	428
4	988	1550	1515

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Problem 2: - Determine the required flow to each recess for a film thickness at the center of the pad of  $h_c = 10 \times 10^{-3}$  in, at a load  $w = 1.5 \times 10^6$  pounds.

To evaluate the flow, we use the performance coefficient relating load, film thickness and flow.

$$\overline{LHQ}_i = \frac{w(h_c)^3}{L^2 q_{ri} \mu}$$

$$q_{ri} = \frac{w(h_c)^3}{(\overline{LHQ}_i)L^2 \mu}$$

Assuming an SAE 30 oil at an operating temperature of 100°F,  $\mu = 15 \times 10^{-6}$  lb-sec/in<sup>2</sup>. Since the same flow is supplied to each recess,

$$q_{ri} = q_r = \frac{1.5 \times 10^6 (10 \times 10^{-3})^3}{(\overline{LHQ}_i)(70)^2 (15 \times 10^{-6})}$$

$$q_r = \frac{20.4}{(\overline{LHQ}_i)}$$

Using the computed values of  $\overline{LHQ}_i$  for the three runs gives,

Run No.	$\overline{LHQ}_i$	$q_r$ , (in <sup>3</sup> /sec)	$q_r$ , (GPM)
7701	1.1196	18.23	4.74
7702	1.4244	14.33	3.73
7703	1.5696	13.00	3.38

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Problem 3: - What will be the film thickness at the center of the pad if the flow for all recesses in all three cases is  $11.53 \text{ in}^3/\text{sec}$  (3 GPM) for a load of  $w = 1.5 \times 10^6 \text{ lbs}$  with  $\mu = 15 \times 10^{-6} \text{ lb-sec/in}^2$ .

Case 7701

$$\frac{w(h_c)^3}{L^2 q_{ri} \mu} = (\overline{LHQ_i}) = 1.1196$$

$$h_c = \left[ \frac{1.1196 (4900)(11.53)(15 \times 10^{-6})}{1.5 \times 10^{+6}} \right]^{1/3}$$

$$h_c = 0.0086 \text{ in.}$$

Case 7702

$$h_c = (h_c \text{ for } 7701) \left[ \frac{\overline{LHQ_i} \text{ for } 7702}{\overline{LHQ_i} \text{ for } 7701} \right]^{1/3} = 0.0086 \left( \frac{1.4244}{1.1196} \right)^{1/3}$$

$$h_c = 0.0093 \text{ in.}$$

Case 7703

$$h_c = (h_c \text{ for } 7701) \left[ \frac{\overline{LHQ_i} \text{ for } 7703}{\overline{LHQ_i} \text{ for } 7701} \right]^{1/3} = 0.0086 \left( \frac{1.5696}{1.1196} \right)^{1/3}$$

$$h_c = 0.0096 \text{ in.}$$

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Problem 4: - What velocity of relative motion will yield a film thickness at the center of 0.010 inch for a  $13.0 \text{ in}^3/\text{sec}$  flow and a load of  $1.5 \times 10^6$  pounds for Run No. 7703.

Using the relationship

$$\frac{\overline{HQ}_i V}{c} = \frac{UL(h_c)}{q_{ri}} = 2.0$$

$$U = \frac{2.0(13.0)}{(70)(10 \times 10^{-3})}$$

$$U = 37.2 \text{ in./sec.}$$

Problem 5: - What is the pad righting moment for all three cases when the load is  $1.5 \times 10^6$  lbs.

To find the pad righting moment,

$$\frac{M}{wL} = (\overline{MW})$$

$$\therefore M = (\overline{MW})wL = (\overline{MW})(1.5 \times 10^6)(70)$$

For the three runs

Run No.	$\overline{MW}$	M(in - lb)
7701	0	0
7702	.1292	$13.6 \times 10^6$
7703	.1306	$13.7 \times 10^6$

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CLOSURE

In the interim between when the above described computer program was first successfully compiled and the writing of this report, a large number of problems have been run. The operating experience compiled has indicated that the program can readily handle a large variety of hydrostatic bearing design and analysis problems. To date, the computer program results have shown good correlation with the results obtained from the electric analog field plotter technique<sup>1</sup> (uniform clearance, zero velocity). It should be mentioned here that in two similar, previously developed hydrostatic bearing computer programs (which allowed the solution of bearing geometries capable of being solved analytically) the correlation between computer program results and known analytical solutions was extremely good.

An experimental program currently underway at The Franklin Institute (being conducted for the Jet Propulsion Laboratory, Contract No. 950-735) is directed toward establishing the correlation between computer program results and those obtainable experimentally (zero velocity, non-uniform film clearance).

In the course of working with the above described computer program, a number of possible additions, modifications and refinements aimed at making the program more efficient and useful have occurred to Jet Propulsion Laboratory and Franklin Institute personnel. They are enumerated below for the record.

1. Input:- The format of the required input information should be changed in order to make key punching and checking of the same much more convenient.
2. Internal Changes:
  - a. The initial value of gamma should be reduced to 1.0 from 1.1.

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- b. Assign a new FORTRAN name to the ratio of pad y-direction length to x-direction length. Change from "AL" to "Y0X". The name AL is confusing since AL(I) is also used as a superscripted variable.
- c. Evaluate the following quantities for subsequent appearance as output.

$$HC = H @ \frac{KK + 1}{2}, \frac{MM + 1}{2}$$

$$ALHCQ(I) = 12(W)(HC)^3/[QQ(I)]$$

$$ALHCV = (\Lambda)/[6(W)(HC)^2]$$

$$HCQV(I) = 2(\Lambda)(HC)/[QQ(I)]$$

$$PRW(I) = [AL(I)]/(W)$$

$$PSW1(I) = [AL(I)]/(W)$$

$$PSW2(I) = \{[QQ(I)]/[FF(I)] + [AL(I)]\}/(W)$$

$$PSW3(I) = (PF)/(W)$$

$$\text{BETA1}(I) = [PRW(I)]/[PSW1(I)]$$

$$\text{BETA2}(I) = [PRW(I)]/[PSW2(I)]$$

$$\text{BETA3}(I) = [PRW(I)]/[PSW3(I)]$$

$$AMXW = (0.50 - CCSI)$$

$$AMYW = [0.50(YOX) - EETA]$$

$$AMW = \{(AMXW)^2 + (AMYW)^2\}^{\frac{1}{2}}$$

$$AMANG = \tan^{-1} \left\{ \frac{(AMYW)}{(AMXW)} \right\}$$

- d. Two recommendations are suggested for reducing the computer time required to obtain the component solutions within subroutine REYN. The first is to initially assign active sill point pressure values other than the currently used value of zero. This would, however, require considerable study before changing. A second possibility is to have

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the program recognize beforehand when identical component solutions for the individual recesses will be obtained. For example a symmetrical distribution of clearance in both the x and y-directions yields identical component solutions for recess combinations 1, 2, 3 and 4 and also for 5 and 6. Tilting of such a symmetrical film distribution about only  $X_1$  produces identical component solutions for recess combinations 1 and 4, 2 and 3, and 5 and 6. Tilting of a symmetrical film distribution about only  $Y_1$  produces identical solutions for recess combinations 1 and 3, and 2 and 4.

3. Output: -

- a. The numerical values of the following quantities should also be written for inclusion in the output:  
YOX,  $X_1$ ,  $Y_1$ , TX, TY, LITER, TRUNC, PLAM, HC, ALHCQ(I), HCQV(I), ALHCV, PRW(I), PSW1(I), PSW2(I), PSW3(I), BETA1(I), BETA2(I) BETA3(I), AMXW, AMYW, AMW, and AMANG.

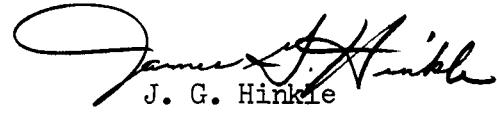
Appropriate titles should also be used for these quantities (See equations [54] through [69]).

- b. Eliminate writing out all "progress indicators". Write out only the last two or three values of PROG(ITER). The run number (NRUN) should also be written out in this portion of the output.

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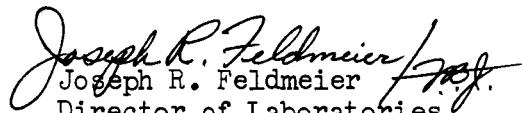
In the judgement of the personnel of The Franklin Institute, all of the goals set forth in our work proposal have been achieved during the course of this program.

  
J. G. Hinkie  
Project Engineer

Approved by:

  
W. W. Shugarts, Jr., Mgr.  
Friction & Lubrication Laboratory

  
N. R. Droulard  
Technical Director

  
Joseph R. Feldmeier  
Director of Laboratories

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3. Shaw, M. C. and E. F. Macks, "Analysis and Lubrication of Bearings," McGraw Hill, 1949, N. Y., N. Y.
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NOMENCLATURE

$AL$	= ratio of pad y-length/pad x-length
$A_n$	= dimensionless clearance coefficients
$E$	= Youngs Modulus (psi)
$H$	= dimensionless film thickness = $h/c$
$K$	= the number of grid cells in the x-direction
$L$	= length of pad in the x-direction, inches
$M$	= the number of grid cells in the y-direction
$M_x$	= moment about $x = \frac{1}{2}$ axis (in-lbs)
$M_y$	= moment about $y = \frac{AL}{2}$ axis (in-lbs)
$N$	= total number of recesses
$P_{ij}$	= dimensionless pressure at grid point $i,j$
$Q_{ij}$	= dimensionless flow out of $i^{th}$ recess corresponding to $j^{th}$ component solution
$QQQ^{(k)}$	= dimensionless flow out of $k^{th}$ pair of opposite recesses
$T_x$	= dimensionless tilt components in x-direction
$T_y$	= dimensionless tilt components in y-direction
$U$	= relative velocity of bearing members, in./sec.
$W$	= dimensionless load
$W_j$	= dimensionless load of the $j^{th}$ component solution
$X$	= dimensionless x-coordinate = $(x/L)$
$Y$	= dimensionless y-coordinate = $(y/L)$
$c$	= a characteristic film thickness, (in.)
$d$	= capillary diameter (in.)
$f_i$	= capillary flow factor for $i^{th}$ recess
$h$	= film thickness, (in.)
$i$	= grid index
$j$	= grid index
$k$	= modulus of the foundation ( $lb/in^2$ ) as defined in reference 4
$l$	= capillary length (in.)
$p$	= pressure, (psia)

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NOMENCLATURE (Cont'd)

- $p_a$  = ambient pressure, (psia)  
 $p^{(f)}$  = dimensionless supply manifold pressure  
 $p_r$  = recess pressure, (psia)  
 $p_{ref}$  = reference pressure, (psia)  
 $p_s$  = supply manifold pressure, (psia)  
 $q_i$  = dimensionless flow out of the  $i^{\text{th}}$  recess  
 $q_{ri}$  = volume flow out of  $i^{\text{th}}$  recess ( $\text{in}^3/\text{sec}$ )  
 $q_{rc}$  = volume flow out of  $i^{\text{th}}$  recess corresponding to  $j^{\text{th}}$  component solution ( $\text{in}^3/\text{sec}$ )  
 $q_{2r}$  = total volume flow out of one pair of opposite recesses ( $\text{in}^3/\text{sec}$ )  
 $s$  =  $X - X_o$   
 $t$  =  $Y - Y_o$   
 $w$  = total load, (lbs)  
 $w_j$  = load carried by  $j^{\text{th}}$  recess, (lbs)  
 $x$  = cartesian coordinate, (in.)  
 $x_j$  = distance in  $x$ -direction to center of pressure of component solution (in.)  
 $y$  = cartesian coordinate, (in.)  
 $y_j$  = distance in  $y$ -direction to center of pressure of component solution (in.)  
 $\Lambda$  = velocity factor =  $\frac{6uUL}{c^2(p_{ref} - p_a)}$   
 $\Lambda' = \Lambda L$   
 $\alpha = \left[ \frac{\Delta X}{\Delta Y} \right]^2$   
 $\alpha_j$  = dimensionless recess pressure  
 $\beta$  = slope of tilted pad in  $h$ ,  $x$ ,  $y$  space  
 $\gamma$  = relaxation factor governing pressure distribution growth  
 $\Delta X = 1/K$   
 $\Delta Y = \frac{AL}{M}$

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NOMENCLATURE (Cont'd)

$\eta$  = dimensionless y coordinate of center of pressure

$\eta_j$  = dimensionless y coordinate of center of pressure for  $i^{\text{th}}$  component solution

$$\lambda = \sqrt{\frac{k}{4 E_1}}$$

$\mu$  = absolute viscosity coefficient (reyns)

$\xi$  = dimensionless x coordinate of center of pressure

$\xi_j$  = dimensionless x coordinate center of pressure for  $i^{\text{th}}$  component solution

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APPENDIX I

PROGRAM BLOCK DIAGRAM

MAIN SUBPROGRAM

Fortran Instruction Listings  
IBM 7094 Compilation Records  
Flow Chart

FLOW SUBPROGRAM

Fortran Instruction Listings  
IBM 7094 Compilation Records  
Flow Chart

REYN SUBPROGRAM

Fortran Instruction Listings  
IBM 7094 Compilation Records  
Flow Chart

FORMH SUBPROGRAM

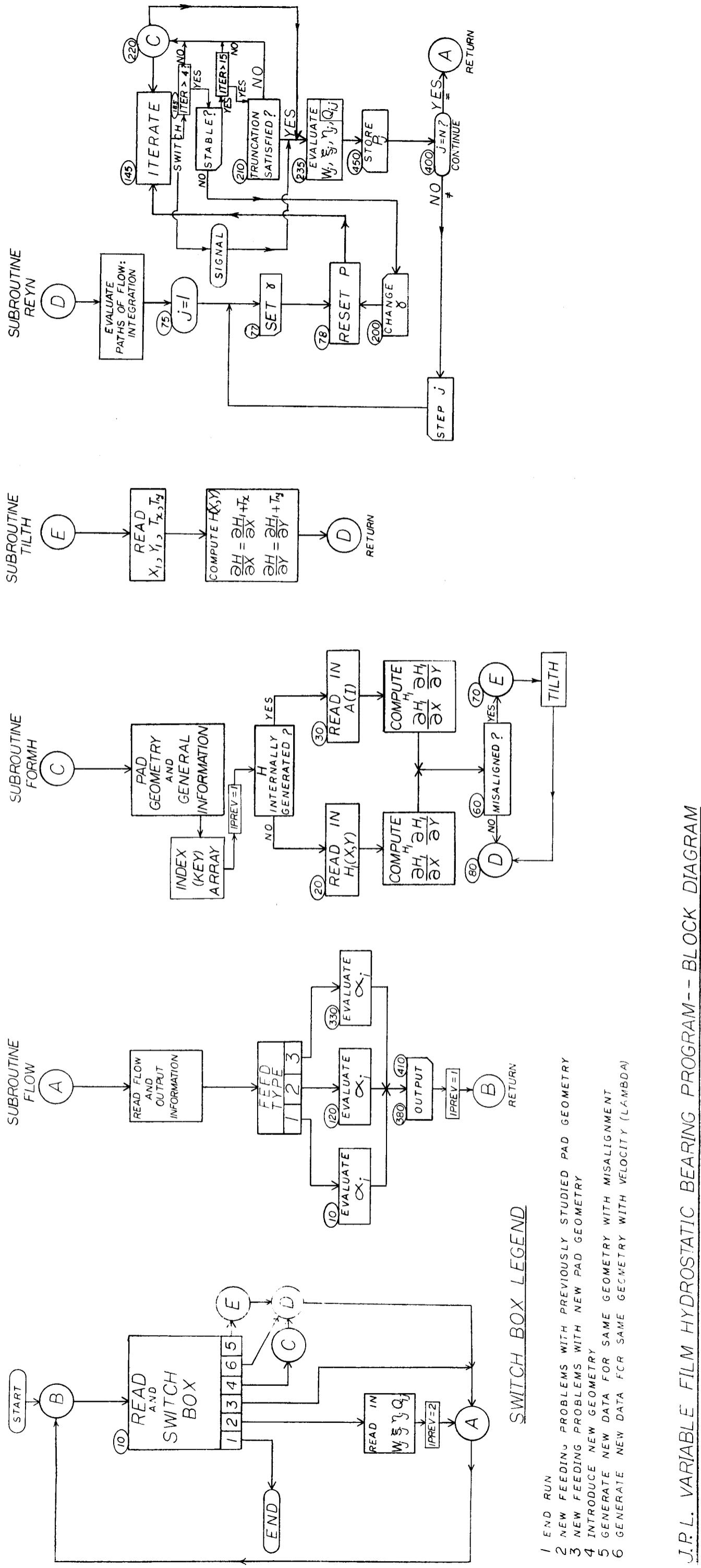
Fortran Instruction Listings  
IBM 7094 Compilation Records  
Flow Chart

TILTH SUBPROGRAM

Fortran Instruction Listings  
IBM 7094 Compilation Records  
Flow Chart

FUNCTION DETER SUBPROGRAM

Fortran Instruction Listings  
IBM 7094 Compilation Records  
Flow Chart



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MAIN PROGRAM FOR HYDROSTATIC BEARINGS WITH VARIABLE FILM THICKNESSES

```

C DEVELOPED FOR J.P.L. JH/VC NOVEMBER, 1962
      DIMENSION H(67,45), HX(67,45), HY(67,45), Q(6,7), W(7), CS(7),
      1 ETA(7), KEY(67,45), P(67,45), A(23)
      COMMON KK, MM, DX, DY, H, HX, HY, K1, K2, K3, M1, M2, TRUNC, Q, LIMJ, M, CSI, ETA,
      1 IPREV, KEY, LITER, P, NIGUAL, PLAM, XI, YI, X, TY, A, INDEX1, INDEX2
      REWIND 15

10 READ INPUT TAPE 5, 1, NSWICH, NIMJ, NIGNAL
      1 FORMAT (11, 212)
      GO TO (20, 30, 40, 50, 60, 70) , NSWICH

20 CONTINUE
      CALL EXIT
      30 READ INPUT TAPE 5, 2, (WL(J), J=1,6), (CSI(J), J=1,6), (((
      1Q(I,J), J=1,6), I=1,NIMJ)
      2 FORMAT (6E12.6)
      LIMJ = NIMJ
      IPREV = 2
      NIGNAL = 0
      CALL FLOW
      GO TO 10
      40 CALL FLOW
      GO TO 10
      50 CALL FORMH
      CALL REYN
      CALL FLOW
      GO TO 10
      60 CALL TILT
      CALL REYN
      CALL FLOW
      GO TO 10
      70 CALL REYN
      CALL FLOW
      GO TO 10
      END(1,0,0,0,0,0,0,0,1,0,0,0,0,0,0,0)
```

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## MAIN PROGRAM FOR HYDROSTATIC BEARINGS WITH VARIABLE FILM THICKNESS

STORAGE NOT USED BY PROGRAM

DEC OCT DEC OCT  
112 00160 17379 41743

## SIMULATIONS LOCATIONS FOR VARIANCES ADDING IN COMMON STATEMENTS

### STORAGE LOCATIONS FOR VARIABLES NOT APPEARING IN COMMON DIMENSION OR EQUIVALENCE STATEMENT

DEC OCT NOV DEC OCT  
111 20157 20158 112 20156

SYMBOLS AND LOCATIONS FOR SOURCE PROGRAM FORMAT STATEMENTS

LOC EFN LOC EFN LOC EFN LOC EFN LOC

## LOCATIONS OF NAMES IN TRANSFER VECTOR

EXIT	4	00004	FLOW	5	00005	FORMH	6	00006	REYN	7	00007	TILT	DEC	OCT
ENTR	5	00005	FRONT	6	00006	FRONT	7	00007	FRONT	8	00008	FRONT	DEC	OCT

SUSTAINABILITY 2020, 12, 3330

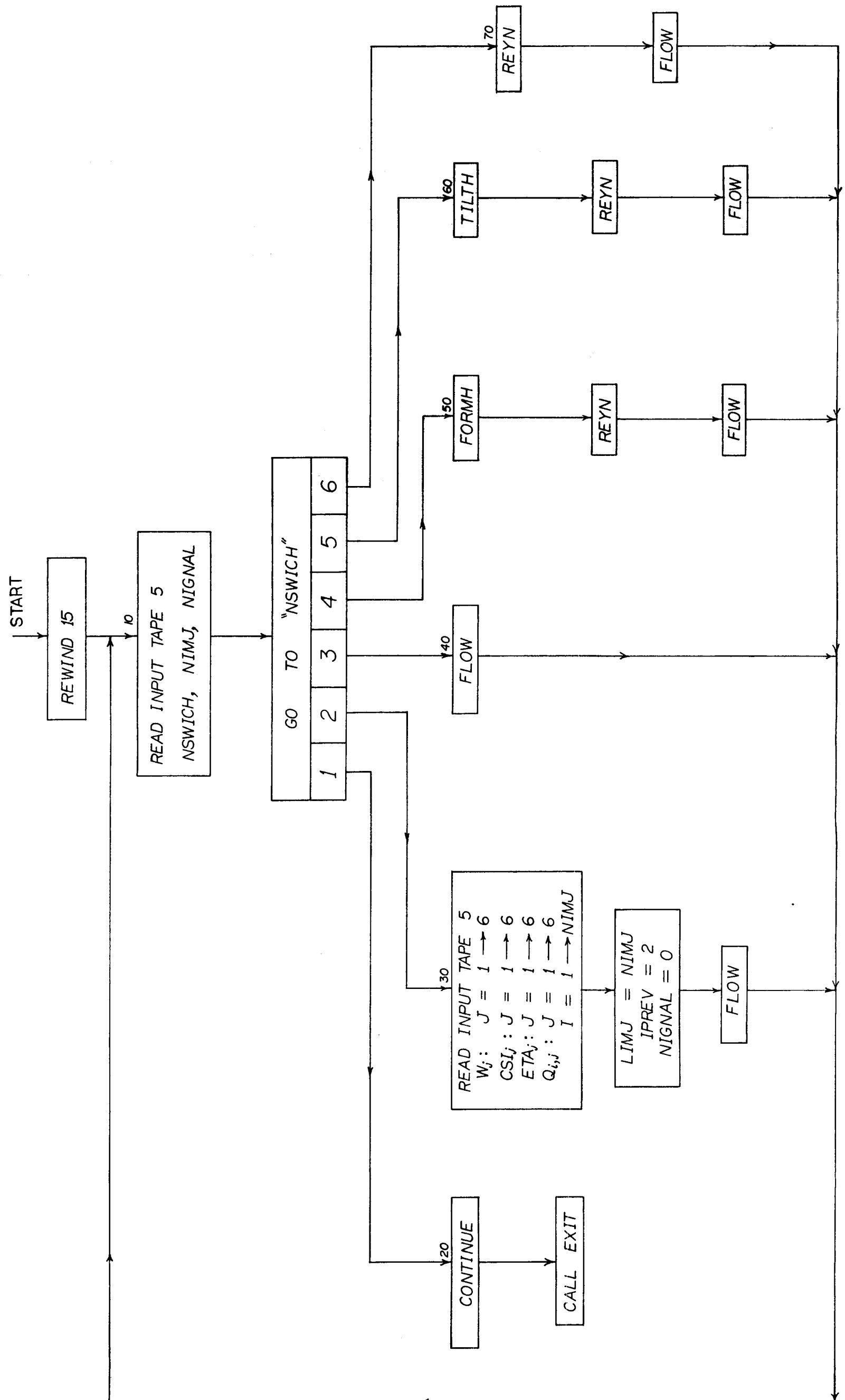
EXTERNAL FORMULA NUMBERS WITH CORRESPONDING INTERNAL FORMULA NUMBERS AND OCTAL LOCATIONS

I-3

MAIN PROGRAM

JET PROPULSION LABORATORY'S VARIABLE FILM HYDROSTATIC BEARING PROGRAM -- BLOCK DIAGRAM

DIMENSION:  $H, HX, HY, Q, W, CSI, ETA, KEY, P, A$   
 COMMON:  $KK, MM, DX, DY, H, HX, HY, K1, K2, K3, M1, M2, TRUNC, Q, LIMJ, W, CSI, ETA, IPREV, KEY, LITER, P, NIGNAL, PLAM, X1, Y1, TX, TY, INDEX1, INDEX2$



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```

SUBROUTINE FLOW
DIMENSION H(67,45), HX(67,45), HY(67,45), Q(6,7), W(7), CSI(7),
1 ETA(7), AL(7), AA(6,7), F(6,6), D(6), QQQ(6), FF(6), PQ(6),
1 P(67,45), PPP(67,45), KEY(67,45), MK(10), T(6,6), TT(6), QQ(6),
1 A(23), B(23), L00(16), NC(23)
COMMON KK,MM,DX,DY,H,HX,HY,K1,K2,K3,M1,M2,TRUNC,Q,LIMJ,M,CSI,ETA,
1 IPREV,KEY,LITER,P,NIGNAL,PLAM,X1,Y1,TX,TY,A,INDEX1,INDEX2
1 FORMAT(15,212)
1 READ INPUT TAPE 5,1 NRUN, NCASE, IOUT
1 IF (NIGNAL - 1) 111, 1112, 111
111 LLLL = LIMJ
GO TO 1115
1112 LLLL = LIMJ + 1
1113 AL(LLLL) = 1.0
1115 GO TO (10, 120, 330), NCASE
10 READ INPUT TAPE 5, 2, (QQ(I,I),III=1,LIMJ)
2 FORMAT (6E12.4)
DO 80 LK = 1,LIMJ
DO 40 I = 1, LIMJ
DO 40 J = 1, LIMJ
IF (J - LK) 30, 20, 30
20 IF (NIGNAL - 1) 26, 27, 26
26 AA(I,J) = QQ(I)
GO TO 40
27 AA(I,J) = QQ(I) - Q(I,LLL)
28 GO TO 40
30 AA(I,J) = Q(I,J)
40 CONTINUE
40 DILK) = DETER(AA,LIMJ)
80 CONTINUE
80 DQ = DETER(Q,LIMJ)
DO 115 LK = 1, LIMJ
115 AL(LK) = DILK) / DQ
GO TO 380
120 LIMJ2 = LIMJ / 2
READ INPUT TAPE 5, 2, (QQQ(I,I),III=1,LIMJ2)
READ INPUT TAPE 5, 2, (FF(I,I), III=1,LIMJ)
DO 130 I = 1, LIMJ, 2
DO 130 J = 1, LIMJ
T(I,J) = Q(I,J) + Q(I+1,J)
130 CONTINUE
130 IF (NIGNAL - 1) 132, 135, 132
132 DO 133 I = 1, LIMJ, 2
133 II = (I+1)/2
133 TT(I) = QQQ(II)
GO TO 139
135 DO 138 I = 1, LIMJ, 2
135 II = (I+1)/2
138 TT(I) = QQQ(II) - Q(I,LIMJ+1) - Q(I+1,LIMJ+1)
139 DO 200 I = 2,LIMJ+2
139 IF (I NIGNAL - 1) 142, 145, 142
142 TT(I) = 0.0
142 GO TO 146
145 TT(I) = Q(I,LIMJ+1)/FF(I) - Q(I-1,LIMJ+1)/FF(I-1)
146 DO 200 J = 1, LIMJ

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      IF (I-J) 140, 140, 150, 140
140  DEL = 0.0
      GO TO 160
150  DEL = 1.0
      160 IF (I-J) 170, 180, 170
170  DDEL = 0.0
      GO TO 190
180  DDEL = 1.0
      190 T(I,J) = (Q(I-1,J)+DEL*FF(I-1))/FF(I)-Q(I,J)+DDEL*F(I)/FF(I)
200  CONTINUE
      205 DO 210 LK = 1, LIMJ
      210 AA(I,J) = TT(I)
      GO TO 230
220  AA(I,J) = T(I,J)
230  CONTINUE
      DILK = DETER(AA,LIMJ)
270  CONTINUE
      DT = DETER (T,LIMJ)
      DO 305 LK = 1, LIMJ
305  AL(LK) = OLIK / DT
      DO 310 I = 1, LIMJ
      QQ(I) = 0.0
      310 QQ(I) = QQ(I) + AL(J) * Q(I,J)
      DO 310 J = 1, LLLL
      310 IF (INCASE - 2) 315, 315, 380
      315 DO 320 I = 1, LIMJ/2
      320 PQ(I)=QQ(2*I-1)/FF(2*I-1) + AL(2*I-1)
      GO TO 380
330  READ INPUT TAPE 5, 5, PF
      5 FORMAT (F10.6)
      READ INPUT TAPE 5, 2, (FF(III), III=1,LIMJ)
      DO 370 I = 1, LIMJ
      370 IF ( NIGUAL - 1) 332, 333, 332
      332 TT(I) = PF * FF(I)
      GO TO 335
      333 TT(I) = PF * FF(I) - Q(I,LIMJ+1)
      335 DO 370 J = 1, LIMJ
      370 IF (I-J) 340, 350, 340
      340 DDEL = 0.0
      GO TO 360
      350 DDEL = 1.0
      360 T(I,J) = Q(I,J) + DDEL * FF(I)
      370 CONTINUE
      GO TO 205
380  WW = 0.0
      DO 390 I = 1, LLLL
      390 WW = WW + AL(I) * W(I)
      CCSI = 0.0
      EETA = 0.0
      DO 400 I = 1, LLLL
      400 CCSI = CCSI + AL(I) * CSI(I)*WW
      EETA = EETA + AL(I) * ETA(I) * W(I) / WW
      IF ( NIGUAL - 1) 402, 403, 402

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SUBROUTINE FLOW
402 WRITE OUTPUT TAPE 6,6,NRUN,LIMJ
403 GO TO 410
403 WRITE OUTPUT TAPE 6,32, NRUN, LIMJ
410 GO TO (40, 430), NCASE
420 WRITE OUTPUT TAPE 6,7
7 FORMAT (31X 58HFEEDING = POSITIVE DISPLACEMENT PUMPS FEEDING EACH
IRECESS. )
GO TO 450
430 WRITE OUTPUT TAPE 6,9
8 FORMAT(14X 92HFEEDING = POSITIVE DISPLACEMENT PUMPS FEEDING PAIRS
10F RECESSES WITH CAPILLARY COMPENSATION. )
GO TO 450
440 WRITE OUTPUT TAPE 6,9,PF
9 FORMAT (15X 36HFEEDING = COMMON CONSTANT PRESSURE (F7.3,6TH ATM.) R
1ESERVOIR FEEDING ALL RECESSES WITH CAPILLARY COMPENSATION. )
/ )
450 IF (NCASE = 2) 452, 451, 452
451 WRITE OUTPUT TAPE 6,15,(QQQ(I), I=1,LIMJ2)
452 FORMAT (11H PUMP FLOWS 3E18.7 )
452 GO TO 460, 4801, IPREV
460 K = KK - 1
M = MM - 1
IF (I(NIGNAL - 1) 454, 453, 454
453 WRITE OUTPUT TAPE 6,28, PLAM
28 FORMAT (10H LAMBDA = F8.4)
29 FORMAT (17H GRID POINTS, K = 13,2X 4H M = 13, 5X 5H K1 = 13, 2X
1 5H K2 = 13, 2X 5H K3 = 13, 3X 5H M1 = 13, 2X 5H M2 = 13
454 WRITE OUTPUT TAPE 6,29, K, M, K1, K2, K3, M1, M2
IF (I(INDEX1 - 1) 455, 459, 455
455 KI = 0
DO 458 L = 1, 23
IF ( ABSF(A(L)) - 0.1E-06) 458, 456, 456
456 KI = KI + 1
B(KI) = A(L)
NC(KI) = L
458 CONTINUE
458 WRITE OUTPUT TAPE 6,4441,(NC(I),B(I)), I = 1,KI)
4441 FORMAT (24H CLEARANCE COEFFICIENTS /5(4H ,AA 12,3H)= E14.8)
4459 IF (INDEX2 - 1) 4460, 461, 4460
4460 IF (INSWICH - 5) 462, 461, 462,
461 WRITE OUTPUT TAPE 6,31, X1, Y1, TX, TY
31 FORMAT (19H TILTED PAD, X1 = E15.8, 2X 6H Y1 = E15.8, 2X 6H TX =
1 E15.8, 2X 6H TY = E15.8 )
462 CONTINUE
GO TO 480
C START A NEW PAGE OF OUTPUT
4620 IF (IGNAL - 1) 463, 464, 463
463 WRITE OUTPUT TAPE 6, NRUN, LIMJ
GO TO 465
464 WRITE OUTPUT TAPE 6,32, NRUN, LIMJ
32 FORMAT (12H1RUN NUMBER 15, 26X 25H HYBRID BEARING PAD WITH 11,
1 10H RECESSES )
465 GO TO ( 466,467,468),NCASE
466 WRITE OUTPUT TAPE 6,7
GO TO 469
467 WRITE OUTPUT TAPE 6,8
GO TO 469

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468 WRITE OUTPUT TAPE 6,9, PF
469 WRITE OUTPUT TAPE 6,3, (KEY(I,I), I = 1, KK)
DO 470 J = 2, MM
470 WRITE OUTPUT TAPE 6,4, (KEY(I,J), I=1, KK )
GO TO 549
3 FORMAT(20H PAD CONFIGURATION. 6711  )
4 FORMAT( 20X 6711  )
480 WRITE OUTPUT TAPE 6,4801
4801 FORMAT( // 54X 10H OUTPUT
IF (LIMJ - 4) .EQ. 490, 490, 500
490 IF (INIGNAL - 1) .EQ. 494, 496, 494
494 WRITE OUTPUT TAPE 6,12, (QQ(I), I=1,4), (Q(I,J), J=1,4)
1(Q(2,J), J=1,4), (CSI(I), I=1,4), (Q(3,J), J=1,4), (ETA(I), I=1,4), Q(4,J),
1), J=1,4)
GO TO 510
12 FORMAT( 8H QQ(I)= 4F9.2,10X10H* Q(I,J)= 4F9.4,8H W(I)= 4F9.4,10
1X2H*8X 4F9.4/8H CSI(I)= 4F9.4,10X 2H*8X 4F9.4/8H ETA(I)= 4F9.4,
110X 2H*8X 4F9.4 )
496 WRITE OUTPUT TAPE 6,112, (QQ(I), I=1,4), (W(I), I=1,5), (CSI(I), I=1,5),
1(ETA(I), I=1,5), (Q(I,J), J=1,5), I=1,4)
GO TO 510
112 FORMAT( 8H QQ(I)= 4F11.4/8H W(I)= 5F9.4/ 8H E
11A(I)= 5F9.4// 10X 5H J=2, 4X 5H J=2, 4X 5H J=4, 4X
1 5H J=5, / 8H Q(I,J)= 5F9.4/ 8H Q(2,J)= 5F9.4/ 8H Q(3,J)= 5F9.4/
1 8H Q(4,J)= 5F9.4 )
500 IF (INIGNAL - 1) .EQ. 504, 504
504 WRITE OUTPUT TAPE 6,13, (QQ(I), I=1,6), (Q(I,J), J=1,6), (W(I), I=1,6),
1,(Q(2,J), J=1,6), (CSI(I), I=1,6), (Q(3,J), J=1,6), (ETA(I), I=1,6), Q(4,J),
1), J=1,6), (Q(5,J), J=1,6), (Q(6,J), J=1,6)
GO TO 510
13 FORMAT( 8H QQ(I)= 6F8.2, 2X 10H* Q(I,J)= 6F8.4/8H W(I)= 6F8.4,2X
1 2H*8X 6F8.4/8H CSI(I)= 6F8.4, 2X 2H*8X 6F8.4/8H ETA(I)= 6F8.4
1,2X 2H*8X 6F8.4/58X 2H*8X 6F8.4/58X 2H*8X 6F8.4/8H E
506 WRITE OUTPUT TAPE 6,113, (QQ(I), I=1,6), (W(I), I=1,7), (CSI(I), I=1,7),
1 (ETA(I), I=1,7), (Q(I,J), J=1,7), (W(I), I=1,7), (CSI(I), I=1,7),
113 FORMAT( 8H QQ(I)= 6F11.4/ 8H W(I)= 7F9.4/ 8H CSI(I)= 7F9.4/ 8H E
11A(I)= 7F9.4// 10X 5H J=2, 4X 5H J=3, 4X 5H J=4, 4X
1 5H J=5, 4X 5H J=6, 4X 5H J=7, / 8H Q(I,J)= 7F9.4/ 8H Q(2,J)= 7F
19.4/ 8H Q(3,J)= 7F9.4/ 8H Q(4,J)= 7F9.4/ 8H Q(5,J)= 7F9.4/
1 8H Q(6,J)= 7F9.4 )
510 IF (INCASE - 2) .EQ. 520, 520
520 WRITE OUTPUT TAPE 6,14, (FF(I), I=1,LIMJ )
14 FORMAT(24H CAPILLARY FACTORS F(I)= 6E16.7 )
530 QQQQ = 0.0
DO 540 I=1,LIMJ
540 QQQQ = QQQQ + QQ(I)
6 FORMAT(12H IRUN NUMBER 15,23X 29HHYDROSTATIC BEARING PAD WITH 11,
1 LOH RECESES. )
WRITE OUTPUT TAPE 6,16,MM,CCSI,EETA,QQQQ
16 FORMAT(28H FINAL RESULTS = TOTAL LOAD= F8.4, 6H CSI=F7.4 , 6H E
11TA= F7.4, 13H TOTAL FLOW= E15.7 )
DO 543 L = 1, LIMJ
543 LOD(L) = L
WRITE OUTPUT TAPE 6,5432, (LOO(L),AL(L), L = 1,LIMJ )
5432 FORMAT(17H RECESS PRESSURES / 2X 6SH, AL( 11, 4H) = F8.5)
GO TO 4620, 549,

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549 IOUT = IOUT + 1
550 GO TO ( 550, 555, 555, 695 ), IOUT
550 IPREV=1
      WRITE OUTPUT TAPE 6,8888
8888 FORMAT (1H1)
      RETURN
17 FORMAT (8E15.8 )
555 GO TO 560,556,IPREV
556 WRITE OUTPUT TAPE 6,5555
5555 FORMAT(52HOPPRESSURE AND CLEARANCE DISTRIBUTIONS NOT AVAILABLE. )
      GO TO 550
560 DO 570 I=1,KK
      DO 570 J=1,MM
570 PPPP(I,J) = 0.0
      REWIND 15
      DD 580 II = 1, LIMJ
      READ TAPE 15,P
      DO 580 I=1,KK
      DO 580 J = 1, MM
580 PPPP(I,J) = PPPP(I,J) + AL(II) * P( I,J )
      REWIND 15
      KK1 = 0
590 IF (KK - KK1 - 10) 600, 600, 610
600 KK2 = KK
      GO TO 620
610 KK2 = KK1 + 10
620 DO 630 L = 1,10
630 MK(L) = KK1 + L
      IF (KK-KK2) 640, 640, 650
640 KK3 = KK2 - KK1
      WRITE OUTPUT TAPE 6, 18, NRUN, (MK(L),L = 1,KK3)
      GO TO 660
650 WRITE OUTPUT TAPE 6, 18, NRUN, (MK(L), L = 1,10 )
660 KK11 = KK1 + 1
670 DO 670 J = 1, MM
670 WRITE OUTPUT TAPE 6, 19,J,(PPPP(I,J),I=KK11,KK2)
19 FORMAT (3H J = 13, 4H, 1OF11.4)
      IF (KK - KK2) 690, 690, 680
680 KK1 = KK1 + 10
      GO TO 590
18 FORMAT ( 1H1 49X 22H PRESSURE DISTRIBUTION//5H RUN 15, 5X 2H I= 13
1,918H I= 13)/)
690 CONTINUE
      IF (IOUT - 2) 550, 695, 550
695 KK1 = 0
700 IF (KK-KK1 - 10) 710, 710, 720
710 KK2 = KK
      GO TO 730
720 KK2 = KK1 + 10
730 DO 740 L = 1, 10
740 MK(L) = KK1 + L
      IF (KK - KK2) 750, 750, 760
750 KK3 = KK2 - KK1
      WRITE OUTPUT TAPE 6, 21, NRUN, (MK(L), L = 1,KK3)
21 FORMAT ( 1H1 49X 22HCLEARANCE DISTRIBUTION //5H RUN 15, 6H I=
113, 9( 8H I= 13)/)

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```
GO TO 770
760 WRITE OUTPUT TAPE 6, 21, NRUN, (MK(L),L=1,10)
770 KK11 = KK1 + 1
    DO 780 J = 1, MM
780 WRITE OUTPUT TAPE 6, 22, J,(H(I,J), I = KK11, KK2)
22 FORMAT (3H J= 13, 4H, 10F11.7 )
IF (KK - KK2) 550, 550, 790
790 KK1 = KK1 + 10
GO TO 700
END(1,0,0,0,0,0,0,0,1,0,0,0,0,0,0)
```

## SUBROUTINE FLOW

## STORAGE NOT USED BY PROGRAM

	DEC	OCT		DEC	OCT
	5109	11765		17379	41743

## STORAGE LOCATIONS FOR VARIABLES APPEARING IN COMMON STATEMENTS

	DEC	OCT		DEC	OCT		DEC	OCT		DEC	OCT									
A	17404	4174	CSI	23456	55640	HX	29542	71546	K1	23512	55730	LIMJ	23464	55650	NIGAL	17410	42002	TX	17406	41776
H	32557	7455							K2	23511	55727									
I	PREV	56522							LITER	20426	47712									
K	32561	77461							PLAM	17409	42001									
M	32560	77460							P	20425	47711									
TRUNC	23507	55723							W	23463	55647									
V1	17407	41777										TY	17405	41775						

## STORAGE LOCATIONS FOR VARIABLES APPEARING IN DIMENSION AND EQUIVALENCE STATEMENTS

	DEC	OCT		DEC	OCT		DEC	OCT		DEC	OCT		DEC	OCT			
AA	5003	11613	AI	5108	11764	B	5055	11677	D	5101	11755	NC	5026	11642	FF	5089	11741
F	4961	11541	LOO	5032	11650	MK	5077	11725	T	1910	03566	QQ	5061	11705	PPP	4925	11475
PQ	5083	11733	QQQ	5095	11747										TT	5067	11713

## STORAGE LOCATIONS FOR VARIABLES NOT APPEARING IN COMMON, DIMENSION, OR EQUIVALENCE STATEMENT

	DEC	OCT		DEC	OCT		DEC	OCT		DEC	OCT		DEC	OCT						
CCSI	1874	03522	DDEL	1873	03521	ICUT	1867	03513	K1	1862	03506	KK2	1857	03501	LLL	1856	03500	L	1855	03477
EETA	1869	03515	II	1868	03514	KK1	1863	03507	LK	1857	03501	NRUN	1852	03474	NSWICH	1851	03473	PF	1856	03472
KI	1864	03510																		
K	1859	03503	LIMJ	1858	03502															
M	1854	03476	NCASE	1853	03475	NN	1848	03470												
QQQQ	1849	03471																		

## SYMBOLS AND LOCATIONS FOR SOURCE PROGRAM FORMAT STATEMENTS

	EFN	LOC		EFN	LOC		EFN	LOC		EFN	LOC		EFN	LOC				
8)1	1	03447	8)2	2	03445	8)3	3	03254	8)4	4	03246	8)5	5	03443				
8)6	6	03014	8)7	7	03441	8)8	8	03425	8)9	9	03403	8)10	10	03235				
8)D	13	03140	8)E	14	03023	8)F	15	03356	8)G	16	02777	8)H	17	02744				
8)1	18	02723	8)J	19	02730	8)L	21	02705	8)M	22	02667	8)N	28	03351				
8)T	29	03345	8)V	31	03310	8)10	32	03271	8)3G	112	03204	8)3H	113	03102				
8)4AP	4441	03321	8)4M1	4801	03242	8)1590	5432	02757	8)5D	5555	02742	8)8BLD	8888	02745				

## LOCATIONS FOR OTHER SYMBOLS NOT APPEARING IN SOURCE PROGRAM

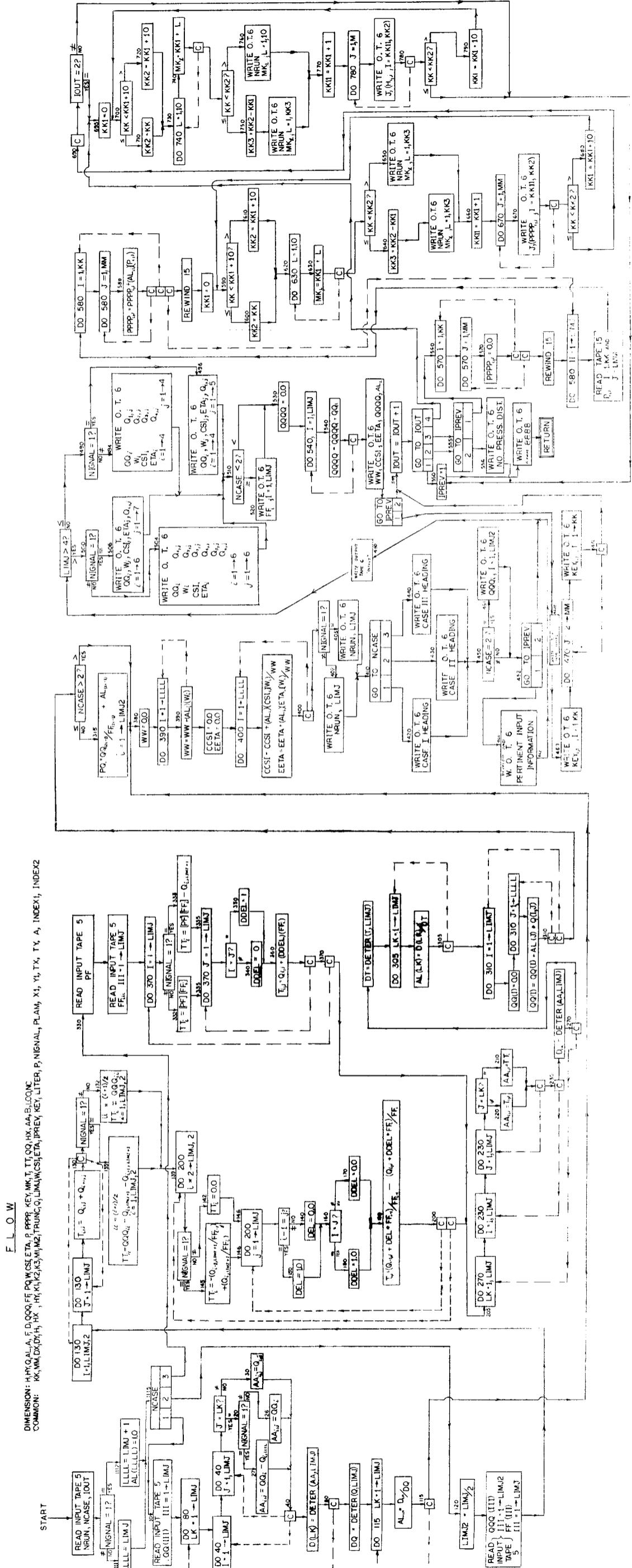
	DEC	OCT																
1)	1832	03450	2)	1440	02640	3)	1450	02652	6)	1453	02655	C)G5	1840	03460	C)60	1836	03454	
C)G1	1837	03455	C)G3	1838	03456	C)G4	1839	03457	C)101	1844	03464	D)11J	1845	03465	C)68	1841	03461	
C)GA	1842	03462	C)GB	1843	03463	C)100	1844	03464	D)114	276	00424	D)27E	1845	03466	C)102	1846	03466	
C)20	1847	03467	D)108	88	00130	D)114	276	00424	D)37E	1324	02454	D)40E	114	00162	D)314	275	00423	
D)130	628	01164	D)23B	716	01314	D)37E	1328	02460	D)40E	114	00162	D)420	580	01104	E)129	492	00754	
D)327	483	00743	D)370	1323	02453	D)714	274	00422	D)727	482	00742							
D)43T	841	01511																

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DETER	(FIL)	LOCATIONS OF NAMES IN TRANSFER VECTOR										ENTRY POINTS TO SUBROUTINES NOT OUTPUT FROM LIBRARY										
		DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	
(SLI)	7 00007	{FIL}	{STH}	4 0004	3 0003	{RLR}	{TSB}	8 00010	6 00006	{RTN}	{TSH}	1 00001	0 00000	{RTN}	{TSH}	5 00005	0 00000	{RTN}	{TSH}	5 00005	0 00000	
<b>EXTERNAL FORMULA NUMBERS WITH CORRESPONDING INTERNAL FORMULA NUMBERS AND OCTAL LOCATIONS</b>																						
111	38 00045	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	
115	53 00146	1112	40 00052	55 00152	27	50 00225	30	57 00157	42 00061	1115	42 00061	10	43 00064	20	52 00143	80	61 00204	133	85 00325	145	95 00433	
115	65 00217	120	67 00225	67 00225	120	90 00370	130	81 00277	132	83 00307	132	93 00430	142	93 00430	145	95 00433	170	102 00464	170	102 00464	210	111 00554
135	87 00333	138	90 00454	98 00454	140	105 00471	190	105 00471	200	106 00505	150	100 00457	160	101 00461	205	107 00522	205	111 00554	210	111 00554	210	111 00554
146	96 00443	220	114 00557	230	114 00561	230	114 00561	270	117 00577	305	121 00612	310	125 00636	310	125 00636	332	139 00750	332	139 00750	333	141 00755	
180	104 00467	315	127 00656	320	128 00662	320	128 00662	330	130 00673	350	146 00774	360	147 00776	360	147 00776	370	148 01002	370	148 01002	402	159 01063	
220	113 00557	335	142 00761	340	144 00771	340	144 00771	350	146 00774	370	157 01050	400	157 01050	402	159 01063	403	162 01074	403	162 01074	440	169 01126	
380	150 01020	380	150 01020	390	152 01025	390	152 01025	400	157 01050	430	167 01117	430	167 01117	440	169 01126	450	171 01140	450	171 01140	453	181 01202	
410	164 01105	410	164 01105	420	165 01110	420	165 01110	430	167 01117	452	177 01165	452	177 01165	460	178 01167	453	183 01214	454	183 01214	454	183 01214	
451	172 01145	451	172 01145	456	189 01262	456	189 01262	458	192 01273	458	192 01273	459	198 01315	4460	199 01320	332	139 00750	332	139 00750	333	141 00755	
455	186 01245	461	200 01325	462	202 01341	462	202 01341	460	204 01342	463	205 01345	463	208 01362	464	208 01362	464	217 01431	464	217 01431	468	215 01417	
465	210 01376	465	210 01376	466	211 01401	466	211 01401	467	213 01410	490	231 01524	494	232 01527	494	232 01527	496	259 01610	496	259 01610	510	332 02017	
470	223 01467	470	223 01467	480	229 01512	480	229 01512	490	231 01524	504	280 01660	506	313 01753	510	332 02017	520	333 02024	520	333 02024	549	351 02133	
500	279 01655	530	338 02043	540	340 02050	540	340 02050	543	344 02102	543	344 02102	543	344 02102	549	351 02133	550	353 02144	550	353 02144	550	353 02144	
555	357 02161	555	358 02164	556	358 02164	556	358 02164	560	360 02173	570	362 02206	570	369 02247	580	369 02247	580	369 02247	580	369 02247	580	369 02247	
590	372 02270	590	373 02275	600	373 02275	600	373 02275	610	375 02300	620	376 02303	620	377 02306	630	377 02306	630	377 02306	630	377 02306	630	377 02306	
640	379 02320	640	387 02345	650	387 02345	650	387 02345	660	393 02364	670	395 02410	670	398 02442	680	402 02442	680	402 02442	680	402 02442	680	402 02442	
690	404 02446	690	406 02455	695	406 02455	695	406 02455	700	407 02462	710	408 02467	710	408 02467	720	410 02472	720	410 02472	720	410 02472	720	410 02472	
730	411 02475	730	412 02500	740	412 02500	740	412 02500	750	414 02512	760	422 02537	760	422 02537	770	428 02556	770	428 02556	770	428 02556	770	428 02556	
780	430 02602	780	437 02634	790	437 02634	790	437 02634															



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SUBROUTINE REYN

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SUBROUTINE REYN
  DIMENSION H(67,45), HX(67,45), HY(67,45), Q(6,7), W(7), CSI(7),
  1 ETA(7), A(23), PPP(67), P(67,45), PROG(1000), KRI(6), KRR(6), MR(6),
  1 MRR(6), KEY(67,45)
  COMMON KK, MM, DX, DY, H, HX, HY, K1, K2, K3, M1, M2, TRUNC, QLIMJ, W, CSI, ETA,
  1 IPREV, KEY, LITER, P, SIGNAL, PLAM, X1, Y1, TX, TY, A, INDEX1, INDEX2
  READ INPUT TAPE 5,1, LITER, TRUNC, PLAM
  1 FORMAT (15, 2E16.8)
  ALPHA = (DX * QX) / (DY * QY)
  FACTOR = 1.0/(2.0*(11.0 + ALPHA))
  K = KK-1
  M = MM-1
  NREL = K + M - K1 - K2 - M1 - M2 - 2
  FK = K
  FM = M
  NACT = (K-1)*(M-1)-(4*(K2+1)+ 2*(K3+1)) *(M2+1)
  PACT = NACT
  IF (K3) 10, 10, 20
  10 LIMJ = 4
  NACT = (K-1)*(M-1) - 4*(K2+1)*(M2+1)
  PACT = NACT
  GC TO 30
  20 LIMJ = 6
  30 KRI(1) = 2
  MR(1) = 2
  MR(3) = 2
  KRP(3) = K
  MRP(14) = M
  KRI(4) = 2
  KRP(14) = 2
  KRR(12) = K
  MRR(12) = M
  IF (LIMJ-4) 50, 50, 40
  40 MR(5) = 2
  MRR(6) = M
  50 MRR(4) = M/2+1
  MRR(1) = M/2+1
  MR(6) = M/2+1
  MRR(5) = M/2+1
  MRR(12) = M/2+1
  MRR(3) = M/2+1
  REWIND 15
  IF (LIMJ -4) 60, 60, 70
  60 KRR(1) = K/2+1
  KRR(3) = K/2+1
  KRR(4) = K/2+1
  KR(2) = K/2+1
  GO TO 71
  70 KKRC = (K-K3+2*K1+2*K2+4)/4
  KRR(1) = KKRC
  KRR(4) = KKRC
  KR(6) = KKRC
  KR(5) = KKRC
  KRR(5) = K - KKRC + 2
  KRR(6) = K - KKRC + 2
  KR(2) = K - KKRC + 2
  KR(3) = K - KKRC + 2

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SUBROUTINE REYN

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71 IF (INIGNAL - 1) 72, 74, 72
72 LLLL = LIMJ
    GO TO 75
74 LLLL = LIMJ + 1, LLLL
75 DO 460 JJJ = 1, LLLL
76 GAMMA = 1.1
78 DO 140 I = 1, KK
    DO 140 J = 1, MM
        IF ( JJJ-LIMJ-1) 79, 80, 80
79 KEYIJ = KEYIJ,J) + 1
    GO TO ( 80, 90, 80), KEYIJ
80 P(I,J) = 0.0
    GO TO 140
90 IF ((I-KR(JJJ)) 80,100, 100
100 IF ((I-KRR(JJJ)) 110,110,80
110 IF ((J - MRR(JJJ)) 80, 120, 120
120 IF ((J-MR(JJJ)) 130, 130, 80
130 P(I,J) = 1.0
140 CONTINUE
145 PGAMMA = 1.0 - GAMMA
    DO 220 LITER = 1, LITER
        PROG(ITER) = 0.0
    DO 146 I = 2,K
146 PPP(I) = P (I,1)
    IF ( JJJ - LIMJ - 1) 147, 1147
147 DO 170 J = 2,M
    DO 170 I = 2,K
        KEYIJ = KEY(I,J) + 1
    GO TO (150, 150, 160), KEYIJ
150 PP = P(I,J)
    GO TO 165
160 PPRIME=FACTOR*(P(I+1,J)+P(I-1,J)+(P(I+1,J)-P(I-1,J))*H(X(I,J))+H(X(I,J))*ALPHA
    1*(P(I,J+1)+P(I,J-1)+P(I,J+1)-P(I,J-1))*HY(I,J))
    PP = GAMMA * PPRIME + PGAMMA * P(I,J)
    PROG(ITER) = PROG(ITER) + PP
165 P(I,J-1) = PPP(I)
    PPP(I) = PP
170 CONTINUE
    GO TO 179
1147 DO 1170 J = 2, M
    DO 1170 I = 2, K
        KEYIJ = KEY(I,J) + 1
    GO TO (1150, 1150, 1160), KEYIJ
1150 PP = P(I,J)
    GO TO 1165
1160 PPRIME=FACTOR*(P(I+1,J)+P(I-1,J)+(P(I+1,J)-P(I-1,J))*(PLAM/1.5)*DX/
    1(H(I,J)*H(I,J))+H(X(I,J)) + ALPHA*(P(I,J+1)+P(I,J-1)-(P(I,J+1)-
    1P(I,J-1))*HY(I,J))
    PP = GAMMA * PPRIME + PGAMMA * P(I,J)
    PROG(ITER) = PROG(ITER) + PP
1165 P(I,J-1) = PPP(I)
    PPP(I) = PP
1170 CONTINUE
1179 DO 1180 I = 2, K
1180 P(I,M) = PPP(I)
    IF ( GAMMA - 0.16) 1181,1181, 1185

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## SUBROUTINE REYN

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181 WRITE OUTPUT TAPE 6,3333, ITER
3333 FORMAT (25H FORCED OUT AT ITERATION 14, 21H, SOLUTION NOT VALID )
GO TO 235
185 IF (ITER - 6) 220, 220, 190
190 IF (PROG(ITER) -2.0*PROG(ITER-2)) 210, 210, 200
200 GAMMA = GAMMA *0.8
WRITE OUTPUT TAPE 6, 1111,JJJ,GAMMA,ITER
1111 FORMAT (7H RECESS 13, 19H, GAMMA CHANGED TO F10.4, 11H, ITERATION
1, 15)
GO TO 78
210 IF (ITER - NREL) 220, 220, 215
215 IF ((PROG(ITER)-PROG(ITER-1))*PACT/(GAMMA*PROG(ITER))-TRUNC) 230,
1, 230, 220
220 CONTINUE
ITER = LITER
230 WRITE OUTPUT TAPE 6,2222,JJJ,ITER,(PROG(LKK), LKK = 1,ITER)
2222 FORMAT (7H RECESS 13, 25H, CONVERGED AT ITERATION 15 / 20H PROGRES
1S INDICATORS / (10F11.6)
235 AAW=0.0
AACSI = 0.0
AAETA = 00.0
J1 = 2
J2 = M
J3 = 2
240 DO 310 J =J1,J2,J3
AW = 0.0
ACSI = 0.0
AAETA = 0.0
FJ = J-1
I1 = 2
I2 = K
I3 = 2
250 DO 260 I = I1, 12, I3
FI = I-1
AW = AW + P(I,J)
ACSI = ACSI + FI*P(I,J)
260 AAETA = AAETA + FJ*P(I,J)
IF (I3 - K) 270, 300, 300
270 AW = 2.0 * AW
ACSI = 2.0 * ACSI
AAETA = 2.0 * AAETA
IF (I1-2) 280, 280, 290
280 I1 = 3
290 I1 = 250
I2 = 1
I3 = K
GO TO 250
300 AAW = AAW + AW
AACSI = AACSI + ACSI
AAETA = AAETA + AEITA
IF ((J3 - M) 320, 350, 350
320 AAW = AAW * 2.0
ACSI = AACS1 * 2.0
AAETA = AAETA * 2.0
IF ((J1-2) 330, 330, 340

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SUBROUTINE REYN

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330 J1 = 3
340 GO TO 240
340 J1 = 1
340 J2 = MM
J3 = M
350 GO TO 240
350 W(JJJJ) = AAW * DX * DY / 9.0
CSI(JJJJ) = AACSI * DX * DY / (9.0*W(JJJJ))
ETA(JJJJ) = AAETA * DX * DY / (9.0*W(JJJJ))
360 DO 440 I11 = 1, LIMJ
QQ = 0.0
J = MR(1111)
I1=KR(1111)
I2=KRR(1111)
370 DO 380 I = 11, 12
380 QQ=QQ-H(I,J)*H(I,J)*H(I,J)*(P(I,J+1)-P(I,J-1))
QQ=QQ+0.5*(H(I,J)*H(I,J)*H(I,J)*H(I,J)*H(I,J)*H(I,J)*
1H(I2,J)*H(I2,J)*(P(I2,J+1)-P(I2,J-1))*
IF (J - MR(1111)) 400, 390, 400
390 QQ = -QQ
J = MRR(1111)
GO TO 370
400 TT = 0.0
I = KR(1111)
J1=MR(1111)
J2=MRR(1111)
410 IF (JJJ-LIMJ- 1) 420, 1420, 1420
420 DO 421 J = J1, J2
421 TT = TT - H(I,J)*H(I,J)*H(I,J)*(P(I1,J) - P(I-1,J))
TT = TT + 0.5*(H(I,J1)*H(I,J2)*H(I,J1)*H(I,J2)*(P(I1+1,J1)-P(I1+1,J1))*
H(I,J2)*H(I,J2)*H(I,J2)*(P(I1+1,J2) - P(I-1,J2)))
425 GO TO 425
1420 DO 1421 J = J1, J2
1421 TT = TT - H(I,J)*H(I,J)*H(I,J)*(P(I1,J) - P(I-1,J)) - 2.0*PLAM*DX
1 * H(I,J)
TT = TT + 0.5*(H(I,J1)*H(I,J1)*H(I,J1)*(P(I1+1,J1) - P(I-1,J1)) +
1 H(I,J2)*H(I,J2)*H(I,J2)*(P(I1+1,J2) - P(I-1,J2)) + 2.0*DX*PLAM*
1*(H(I,J2) + H(I,J1)))
425 IF (I - KR(1111)) 440, 430, 440
430 TT = - TT
I = KRR(1111)
GO TO 410
440 Q(1111,JJJ) = QQ*DX*0.5/DY+TT*DY*0.5/DX
450 WRITE TAPE 15,P
2 FORMAT (8E15.8)
460 CONTINUE
END FILE 15
REWIND 15
RETURN
END(1,0,0,0,0,0,0,1,0,0,0,0,0)
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SUBROUTINE REYN

STORAGE NOT USED BY PROGRAM

DEC OCT  
2418 04562

17379 41743

STORAGE LOCATIONS FOR VARIABLES APPEARING IN COMMON STATEMENTS

	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT
A	17404	41774	CSI	23559	77457	DY	32558	77456	ETA	23449	55631	
H	32557	77455	HX	26527	63637	INDEX1	17381	41745	INDEX2	17380	41744	
IPREV	23442	55622	K1	25512	55727	K3	23510	55726	KEY	23441	55621	
KK	32561	77461	LIMJ	24464	55650	LITER	20426	47712	M1	23509	55725	
MM	32560	77460	NIGNAL	17410	42001	PLAM	17409	42001	P	20425	47711	
TRUNC	23507	55723	TX	17406	41776	TY	17405	41775	W	23463	55647	
Y1	17407	41777							X1	17408	42000	

STORAGE LOCATIONS FOR VARIABLES APPEARING IN DIMENSION AND EQUIVALENCE STATEMENTS

	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	
KRR	1344	02500	KR	1350	02506	MRR	1332	02464	MR	1338	02472
PROG	2350	04456							PPP	2417	04561

STORAGE LOCATIONS FOR VARIABLES NOT APPEARING IN COMMON, DIMENSION, OR EQUIVALENCE STATEMENT

	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	
AACSI	1326	02456	AAETA	1325	02455	AAW	1324	02454	ACSI	1323	02453
ALPHA	1321	02451	AW	1320	02450	FACTOR	1319	02447	FI	1318	02446
FK	1316	02444	FM	1315	02443	GAMMA	1314	02442	I1	1313	02441
I3	1311	02437	I	1310	02436	ITER	1309	02435	J1	1308	02434
J3	1306	02432	JJJ	1305	02431	J	1304	02430	KEYIJ	1303	02427
K	1301	02425	LLL	1300	02424	M	1299	02423	NACT	1298	02422
PACT	1296	02420	PGAMMA	1295	02417	PPRIME	1294	02416	PP	1293	02415
TT	1291	02413							QQ	1292	02414

SYMBOLS AND LOCATIONS FOR SOURCE PROGRAM FORMAT STATEMENTS

	EFN	LOC	EFN	LOC	EFN	LOC	EFN	LOC						
8)1	1	02365	8)2	2	02316	8)12N	1111	02350	8)125E	2222	02335	8)1385	3333	02362

LOCATIONS FOR OTHER SYMBOLS NOT APPEARING IN SOURCE PROGRAM

	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	DEC	OCT	
IJ	1270	02366	2)	1205	02285	3)	1214	02276	6)	1223	02307
A1104	1166	02216	A)105	1179	02233	A)106	1192	02250	C)1G1	1278	02376
C)1G3	1280	02400	C)1G6	1281	02401	C)100	1282	02402	C)103	1283	02403
C)105	1285	02405	C)106	1286	02406	C)107	1287	02407	C)204	1288	02410
C)1206	1290	02412	D)115	689	01261	D)1200	3115	00473	D)1212	469	00725
D)121M	641	01201	D)210	648	01201	D)40C	311	00467	D)40H	347	00533
E)1G	337	00521	E)M	373	00565	E)11	435	00663	E)13	477	00735
E)1B	556	01054	E)123	775	01407	E)10H	349	00535	E)20H	349	00535

LOCATIONS OF NAMES IN TRANSFER VECTOR

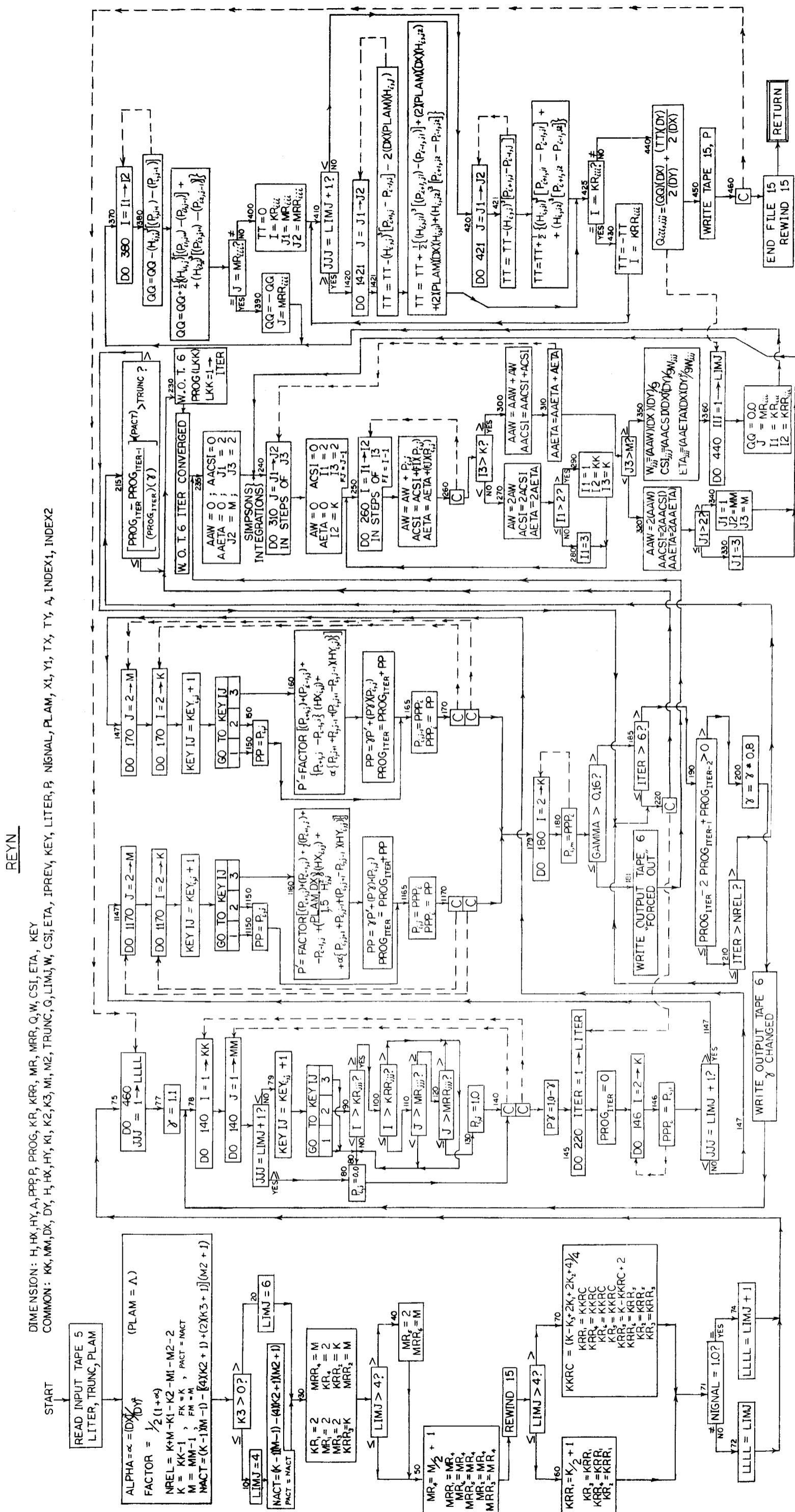
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SUBROUTINE REYN

(LEFT)	DEC	OCT	(FIL) (STB)	DEC	OCT	(RTN) (TSB)	DEC	OCT	(RTN) (WLR)	DEC	OCT	(SLD)	
ENTRY POINTS TO SUBROUTINES NOT OUTPUT FROM LIBRARY													
(LEFT)	(FIL)	(RTN)	(RHT)	(SLD)	(STB)	(STH)	(LTH)	(STH)	(LTH)	(WLR)			
EXTERNAL FORMULA NUMBERS WITH CORRESPONDING INTERNAL FORMULA NUMBERS AND OCTAL LOCATIONS													
EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	IFN	LOC
10	22	00153	20	26	00212	30	27	00214	40	36	00241	50	38
60	46	00333	70	51	00374	71	60	00444	72	61	00447	74	63
75	64	00455	77	65	00470	78	66	00474	79	69	00523	80	71
90	73	00542	100	74	00550	110	75	00554	120	76	00561	130	77
140	78	00571	145	80	00605	146	83	00622	147	85	00634	150	89
160	91	00665	165	94	00726	170	96	00732	1147	98	00743	1150	102
1160	104	00774	1165	107	01045	1170	109	01051	1179	110	01061	180	111
181	113	01106	185	116	01121	190	117	01126	200	118	01137	210	122
215	123	01166	220	124	01202	230	126	01211	235	132	01234	240	138
250	146	01314	260	150	01353	270	152	01367	280	156	01404	290	158
300	162	01417	310	164	01425	320	166	01437	330	170	01454	340	172
350	176	01474	360	179	01526	370	184	01562	380	185	01576	390	188
400	191	01665	410	195	01707	420	196	01714	421	197	01730	1420	200
1421	201	02013	425	203	02103	430	204	02111	440	207	02124	450	208
460	210	02156											

DIMENSION : H, HX, HY, A, PPP, PROG, KR, KRR, MR, MRR, Q, W, CSI, ETA, K  
 COMMON : KK, MM, DX, DY, H, HX, HY, K1, K2, K3, M1, M2, TRUNC, Q, LIM, J, W, CS



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SUBROUTINE FORMH
  DIMENSION H(67,45), HX(67,45), HY(67,45), A(23), KEY(67,45),
  1 Q(6,7), W(7), CSI(7), ET(7), P(67,45)
  COMMON KK, MM, DX, DY, H, HX, HY, K1, K2, K3, M1, M2, TRUNC, Q, LIMJ, MCSI, ETIA,
  1 IPREV, KEY, LITER, P, NIGNAL, PLAM, XI, VI, TX, TY, A, INDEX1, INDEX2
  1 FORMAT(19I3, E16.8)
  READ INPUT TAPE 5, 1, K, M, INDEX1, INDEX2, K1, K2, K3, M1, M2, AL
  KK = K + 1
  MM = M + 1
  FK = K
  FM = M
  DX = 1.0/FK
  DY = AL/FM
  FX = 0.5/DX
  FY = 0.5/DY
  DO 11 J = 2, M
  KEY(I,J) = 0
  KEY(KK,J) = 0
  DO 11 I = 2, K
  11 KEY(I,J) = 1
  DO 21 I = 1, KK
  KEY(I,I) = 0
  21 KEY(I,MM) = 0
  KK1 = K1 + K2 + 2
  KK2 = (K - K3)/2
  DO 51 J = 2, M
  DO 31 I = 2, K1
  I1 = K + 2 - I
  KEY(I,J) = 2
  31 KEY(I1,J) = 2
  DO 41 I = 1, KK1, KK2
  I1 = K + 2 - I
  KEY(I,J) = 2
  41 KEY(I1,J) = 2
  IF (K3) 42, 42, 51
  42 I1 = K/2 + 1
  KEY(I1,J) = 2
  51 CONTINUE
  MM2 = M - M1 - M2
  MM1 = M1 + M2 + 2
  DO 81 I = 2, K
  DO 61 J = 2, M1
  JJ = M + 2 - J
  KEY(I,J) = 2
  61 KEY(I,J) = 2
  DO 71 J = MM1, MM2
  71 KEY(I,J) = 2
  81 CONTINUE
  IPREV = 1
  2 FORMAT (6E11.9)
  IF (INDEX1 - 1) 30, 20, 30
  20 READ INPUT TAPE 5, 2, ((H(I,J), I=1, KK), J=1, MM)
  DO 25 I = 2, K
  DO 25 J = 2, M
  HX(I,J) = ((H(I+1,J)-H(I-1,J))*75/H(I,J))
  25 HY(I,J) = ((H(I,J+1)-H(I,J-1))*75/H(I,J))

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SUBROUTINE FORMH
3 FORMAT (5E14.8)
GO TO 60
30 READ INPUT TAPE 5,3,(A(L),L = 1,23)
CON = EXP(-5.0*A(21))*COSF(A(21)*0.5) * 2.0
DO 50 I = 1,KK
F1 = 1
X = (F1 - 1.0) * DX
S = (F1 - 1.0) * DX - A(22)
SS = S * S
DO 50 J = 1, MM
FJ = J
T = (FJ - 1.0) * D0 - A(23)
TT = T * T
TT = TT * T
H(I,J)=A(11)+A(2)*S +A(3)*T+A(4)*SS+A(5)*TT+A(6)*S*T+A(7)*SS+T+A(8)*
1TTT+A(9)*SS*T+A(10)*S*TT+A(11)*SQRTRF(A(12)+A(13)*SS*(A(14)*TT)+
1*(A(15)*COSF(A(16)*S)+A(17)*COSFA((18)*T)+A(19)*COSFA((16)*S)*COSF(
1A(18)*T)-A(20)*(EXP(-A(21)*X)*COSF(A(21)*X)+EXP(-A(21)*X)*COSF(A(21)*
1COSF(A(21)*X)-(1.0-X)*CON)
H(X,I,J) =(A(12)+2.*A(4)*S+A(6)*T+3.*A(7)*SS+2.*A(9)*S*T+A(10)*TT-
1A(16)*SINF(A(16))+S*(A(15)+A(19)*COSF(A(18)*T)))*1.5*D/X/H(I,J)+A(
120)*EXP(-A(121)*X)*(A(21)*X)+SINF((A(21)*X)+SINF((A(21)*X))-EXP(-A(
121)*X)*(A(21)*X)+SINF((A(21)*X)+(1.0-X)*SINF((A(21)*X)+(1.0-X))))*1.5*D/X/H(I,
1,J)
HY(I,J)=(A(11)+2.*A(5)*T+A(6)*S+3.*A(8)*TT+A(9)*SS-A(11)*SINF(A(18)*
1*T)*(A(17)+A(19)*COSF(A(16)*S)),1.5*D/Y/H(I,J))
SL = A(121)*A(13)*SS*(A(14)*TT
1F (SL) 50, 50, 40
4.0 SL = A(11)*SQRTRF(SL)
HX(I,J) = HY(I,J)+SL*A(13)*S+1.5*D/X/H(I,J)
HY(I,J) = HY(I,J) + SLL*A(14)*T+1.5*D/Y/H(I,J)
50 CONTINUE
60 IF (INDEX2 - 1) = 80, 70, 80
70 CALL TILT
END

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SUBROUTINE FORMH

DEC OCT  
941 01655

DEC OCT  
17379 41743

STORAGE NOT USED BY PROGRAM

	DEC	OCT	DEC	OCT
A	17404	41774	CSI	23456 55640
H	32557	77455	MX	29542 71546
I	PREV	23442 55622	K1	23512 55730
KK	32561	77461	LIMJ	23464 55650
MM	32560	77460	NIGNAL	17410 42002
TRUNC	23507	55723	TX	17406 41776
Y1	17407	41777		

STORAGE LOCATIONS FOR VARIABLES APPEARING IN COMMON STATEMENTS

	DEC	OCT	DEC	OCT
AL	940	01654	CON	939 01653
FM	935	01647	FX	934 01646
JJ	930	01642	J	929 01641
MM1	925	01635	MM2	924 01634
S	920	01630	SS	919 01627
TTT	915	01623	X	914 01622

STORAGE LOCATIONS FOR VARIABLES NOT APPEARING IN COMMON, DIMENSION, OR EQUIVALENCE STATEMENT

	DEC	OCT	DEC	OCT
			F1	938 01652
			FY	933 01645
			KK1	928 01640
			M	923 01633
			SSS	918 01626
			X	914 01622

SYMBOLS AND LOCATIONS FOR SOURCE PROGRAM FORMAT STATEMENTS

	EFN	LOC	EFN	LOC	EFN	LOC
811	EFN 1	01552	812	2 01550	813	3 01546

LOCATIONS FOR OTHER SYMBOLS NOT APPEARING IN SOURCE PROGRAM

	DEC	OCT	DEC	OCT	DEC	OCT
1)	875	01553	2)	850 01522	3)	81531
A)106	837	01505	C)162	903 01607	C)100	904 01610
C)105	907	01613	C)106	908 01614	C)200	909 01615
C)206	912	01620	C)208	913 01621	D)20A	194 00302
D)411	436	00664	E)C	215 00327		

LOCATIONS OF NAMES IN TRANSFER VECTOR

	DEC	OCT	DEC	OCT	DEC	OCT
COS (RTN)	2 00002	1 00001	EXP (RTN)	3 00003 0 00000	SIN (RTN)	5 00005

ENTRY POINTS TO SUBROUTINES NOT OUTPUT FROM LIBRARY

CUS	EXP	SIN	SQRT	WILTH	(RTN)	(TSW)
-----	-----	-----	------	-------	-------	-------

EXTERNAL FORMULA NUMBERS WITH CORRESPONDING INTERNAL FORMULA NUMBERS AND OCTAL LOCATIONS

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SUBROUTINE FORMH			SUBROUTINE FORMH		
EFN	IFN	LOC	EFN	IFN	LOC
11	22 00141	21	25 00172	31	32 00263
51	40 00346	61	47 00467	71	49 00477
25	63 00606	30	65 00621	40	87 01407
70	92 01463	80	93 01464		

SUBROUTINE FORMH

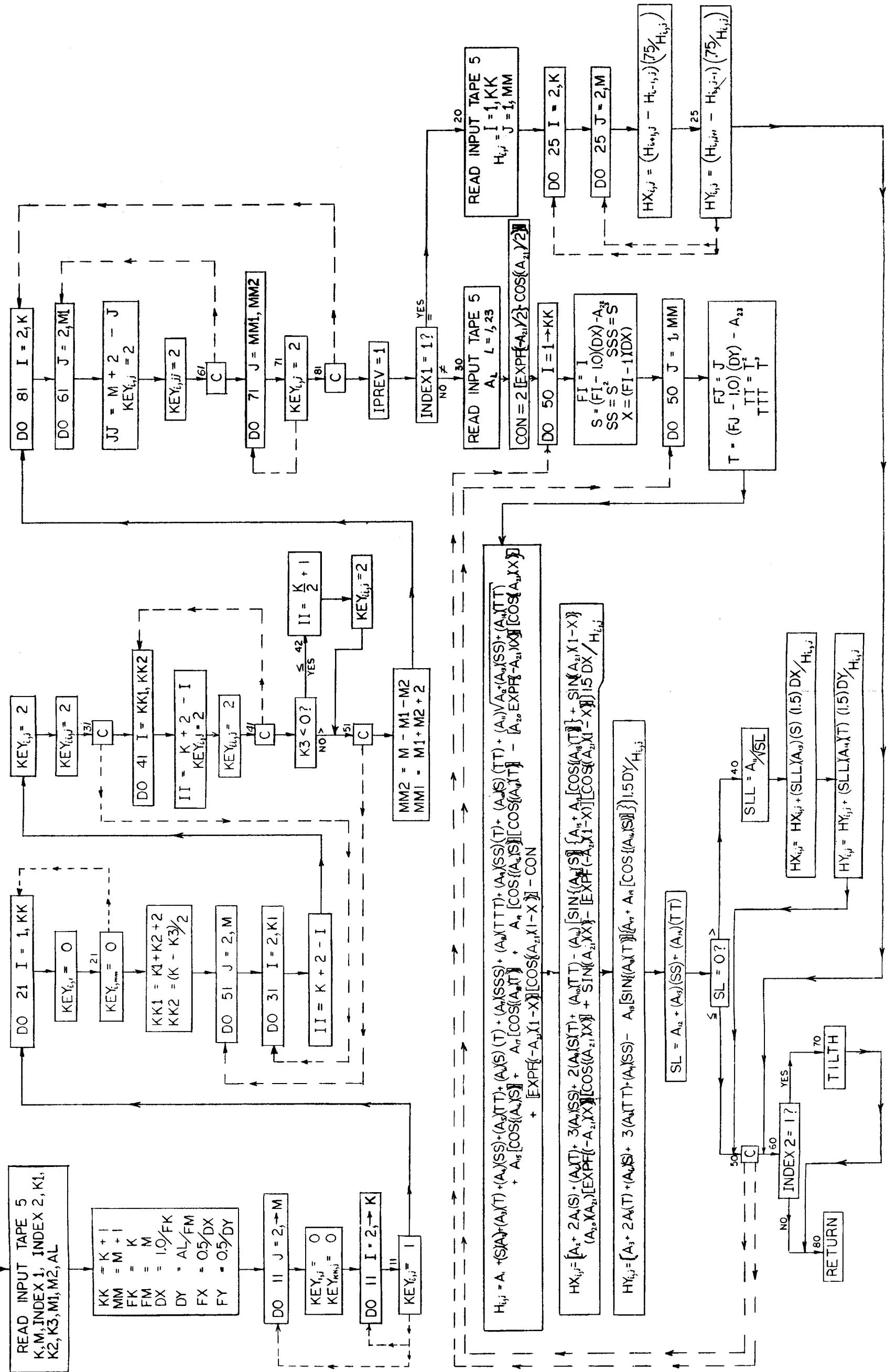
EFN	IFN	LOC	EFN	IFN	LOC
36 00315	41	36 00315	42	38 00330	
50 00504	81	50 00504	20	53 00526	
50 01443	50	90 01443	60	91 01456	

FORM H

```

DIMENSION: H, HX, HY, A, KEY, Q, W, CSI, ETA, P
COMMON: KK, MM, DX, DY, H, HY, K1, K2, K3, M1, M2, TRUNC, Q, LIMJ, W, CSI, ETA, IPREV, KEY, LITER, R, SIGNAL, PLAM, X1, Y1, TX, TY, INDEX1, INDEX2

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SUBROUTINE FILTH
  DIMENSION H(67,45), HX(67,45), HY(67,45), KEY(67,45), Q(6,7),
  1, CSI(7), ETA(7), A(23), P(67,45)
  COMMON KK,MM,DX,DY,HX,HY,K1,K2,K3,M1,M2,TRUNC,Q,LINJ,W,CSI,ETA,
  1,IPREV,KEYLITER,P,NIGHT,PLAMM,X1,Y1,TX,TY,A,INDEX1,INDEX2
  READ INPUT TAPE 5, 1, X1, Y1, TX, TY
  FORMAT (4E15.8)
  DO 90 I = 1, KK
    F1 = I
    X = (F1 - 1.0)* 0X - X1
    DO 90 J = 1, MM
      FJ = J
      H(I,J) = H(I,J)*TX*X*TY*((FJ-1)*DY-YI)
      HX(I,J) = HX(I,J) + TX * 1.5 * DX / H(I,J)
      HY(I,J) = HY(I,J) + TY * 1.5 * DY / H(I,J)
      CONTINUE
  90 RETURN
  END

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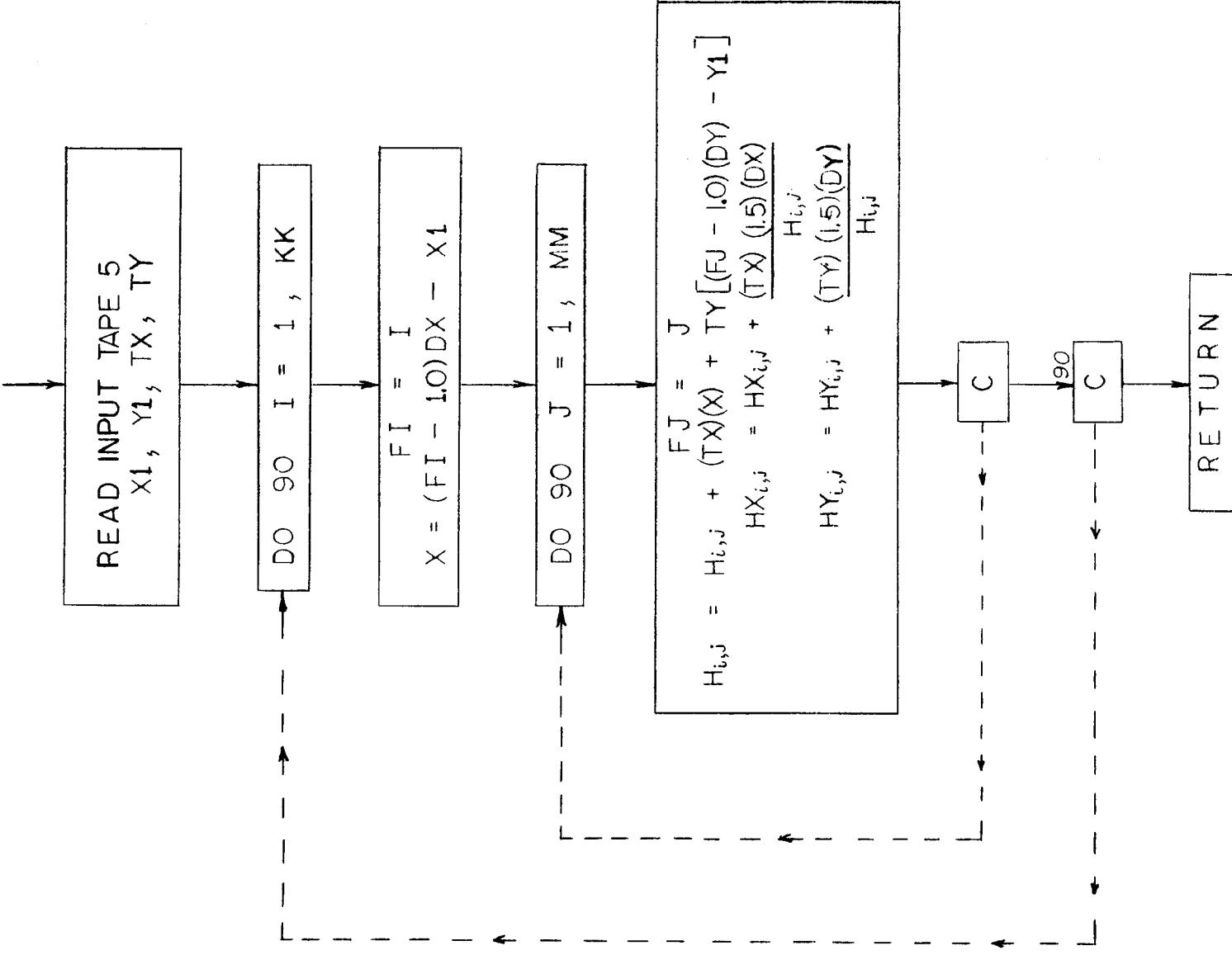
F-B2099

SUBROUTINE	MAIN	STORAGE NOT USED BY PROGRAM			
		DEC	OCT	DEC	OCT
		115	00163	17379	41743
STORAGE LOCATIONS FOR VARIABLES APPEARING IN COMMON STATEMENTS					
A	DEC OCT	DEC OCT	DEC OCT	DEC OCT	DEC OCT
H	17404 41774	CSI 23456 55640	DX 32559 77457	DY 32558 77456	ETA 23449 55631
I	32557 77455	HX 29542 71546	HY 26527 63637	INDEX1 17381 41745	INDEX2 17380 41744
PREV	23442 55622	K1 23512 55730	K2 23511 55727	K3 23510 55726	KEY 23441 55621
KK	32561 77461	LIMJ 23464 55650	LITER 20426 47712	M1 23509 55725	M2 23508 55724
MM	32560 77460	NIGMJ 17410 42002	PLAM 17409 42001	P 20425 47711	Q 23506 55722
TRUNC	23507 55723	TX 17406 41776	TY 17405 41775	W 23463 55647	X1 17408 42000
YI	17407 41777				
STORAGE LOCATIONS FOR VARIABLES NOT APPEARING IN COMMON, DIMENSION, OR EQUIVALENCE STATEMENT					
FI	DEC OCT	DEC OCT	DEC OCT	DEC OCT	DEC OCT
	114 00162	FJ 113 00161	I 112 00160	J 111 00157	X 110 00156
SYMBOLS AND LOCATIONS FOR SOURCE PROGRAM FORMAT STATEMENTS					
8)1	EFN LOC	EFN LOC	EFN LOC	EFN LOC	EFN LOC
LOCATIONS FOR OTHER SYMBOLS NOT APPEARING IN SOURCE PROGRAM					
1)	DEC OCT	DEC OCT	DEC OCT	DEC OCT	DEC OCT
	106 00152	2) 93 00135	3) 96 00140	6) 98 00142	
LOCATIONS OF NAMES IN TRANSFER VECTOR					
(RTN)	DEC OCT	DEC OCT	DEC OCT	DEC OCT	DEC OCT
	1 00001	(TSH) 0 00000			
ENTRY POINTS TO SUBROUTINES NOT OUTPUT FROM LIBRARY					
(RTN)	(TSH)				
EXTERNAL FORMULA NUMBERS WITH CORRESPONDING INTERNAL FORMULA NUMBERS AND OCTAL LOCATIONS					
EFN	IFN LOC	EFN IFN LOC	EFN IFN LOC	EFN IFN LOC	EFN IFN LOC
90	16 00120				

# TILT

JPL VARIABLE FILM HYDROSTATIC BEARING PROGRAM - BLOCK DIAGRAM

DIMENSION : H, HX, HY, A, KEY, Q, W, CSI, ETA, A, P  
 COMMON : KK, MM, DX, DY, H, HX, HY, ... X<sub>I</sub>, Y<sub>I</sub>, TX, TY, A, INDEX1, INDEX2  
 START



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FUNCTION DETER(AA,LIMJ).

STORAGE NOT USED BY PROGRAM

	DEC	OCT		DEC	OCT
DETER	264 00410			32561 77461	
M	258 00402				

STORAGE LOCATIONS FOR VARIABLES NOT APPEARING IN COMMON, DIMENSION, OR EQUIVALENCE STATEMENT

	DEC	OCT		DEC	OCT
DETER	263 00407	1		262 00406	J
M	258 00402	N		257 00401	PROD

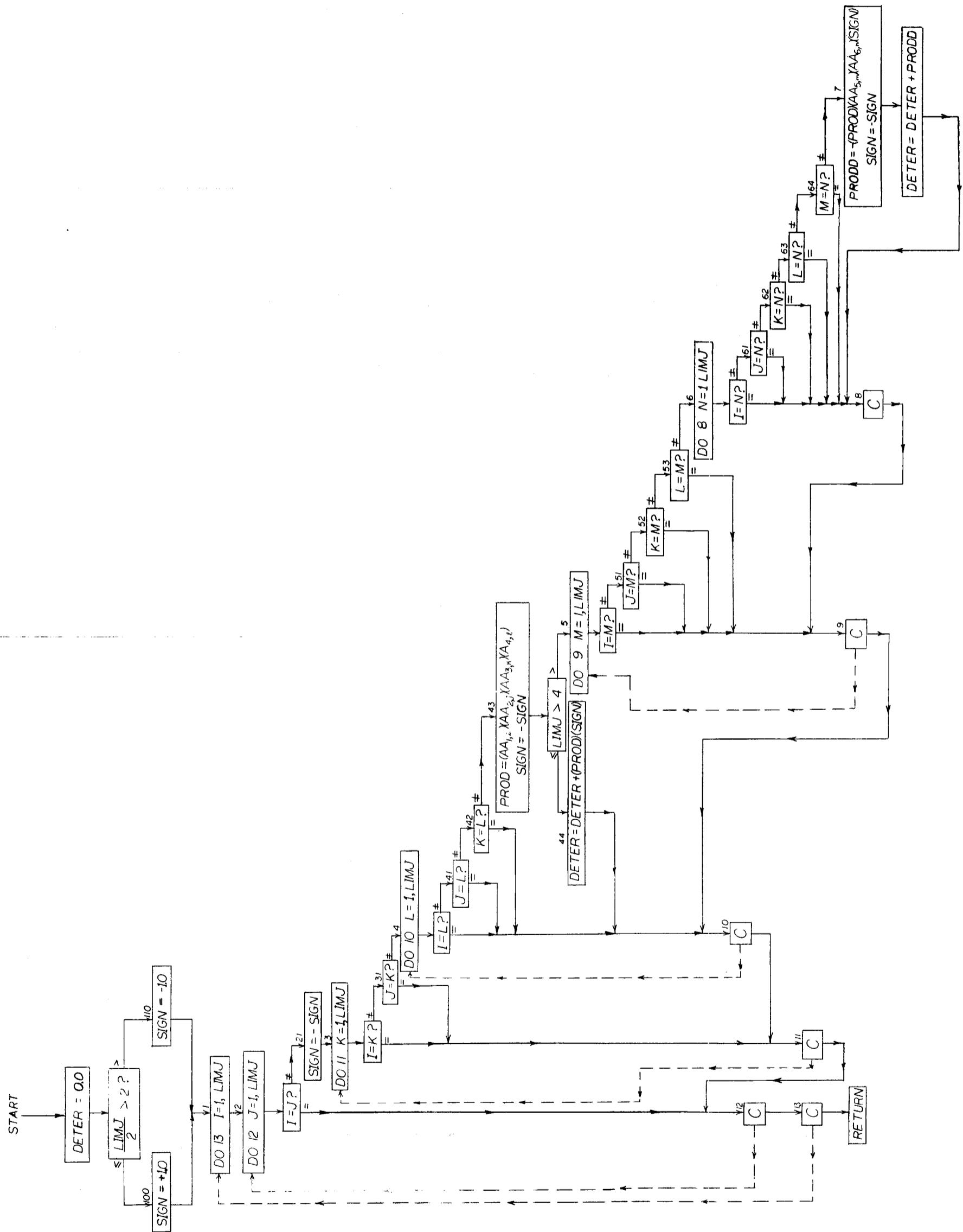
LOCATIONS FOR OTHER SYMBOLS NOT APPEARING IN SOURCE PROGRAM

	DEC	OCT		DEC	OCT
C101	227 00343	3)		230 00346	6)
C1200	243 00363	C102		244 00364	C103
C1205	248 00370	C1201		249 00371	C1202
D140V	253 00375	D1105		63 00077	D1107
E1S	183 00267	D160V		192 00300	E1C
E1111	199 00307	E1T		187 00273	E111
		E120R		176 00260	E130T

EXTERNAL FORMULA NUMBERS WITH CORRESPONDING INTERNAL FORMULA NUMBERS AND OCTAL LOCATIONS

	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC	EFN	IFN	LOC
100	6 00052	110	8 00055	1	9 00057		2	10 00067		21	13 00103	
3	14 00105	31	16 00121	4	17 00124		41	19 00135		42	20 00143	
43	21 00146	44	24 00165	5	26 00172		51	28 00204		52	29 00207	
53	30 00212	6	31 00215	61	33 00226		62	34 00231		63	35 00234	
64	36 00237	7	37 00242	8	40 00263		9	41 00274		10	42 00302	
11	43 00310	12	44 00316	13	45 00326							

FUNCTION      DETER  
DIMENSION: AA



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APPENDIX II

IBM 7094 LOADING RECORD

(Six Subprograms)

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ENTRY POINTS TO SUBROUTINES REQUESTED FROM LIBRARY,	(RTN) (RWT) (STB)	(TSHM) (WLR) (SLO)	(RN) (EF1)	EXIT COS	(STHM) EXP	(FIL) SQRT	(TSB) SIN	(SLI)	(RLR)
ENTRY NAME	ENTRY ADD.		NO. OF TRANSFERS	LOAD ADD.					
MAIN	00155		00011	00144					
FLOW	00341		00011	00324					
REYN	12326		00011	12311					
FILTH	17101		00002	17073					
FORMH	17271		00007	17256					
DETER	21137		00000	21133					
ERRDMP	21544		00001	21543					
EXIT	21547								
ITES	21571		00000	21571					
(FPT)	21572		00000	21572					
EXP	21630		00000	21630					
SIN	21710		00000	21707					
COS	21707								
SQRT	22060		00000	22060					
(RLR)	22221		00011	22134					
(TSB)	22145		00007	22236					
(WLR)	22245		00006	22323					
(STB)	22334		00006	22323					
(TSHM)	22331		00006	22360					
(TSHD)	22401								
(STHM)	22371								
(STH)	22366		00002	22510					
(EFT)	22512		00002	22517					
(RW)	22521		00000	22542					
(SLI)	22542		00000	22557					
(SLO)	22557		00007	22514					
(RDC)	22625								
(RER)	22603		00012	22641					
(WTC)	22704								
(WER)	22653		00001	22734					
(SET)	23011								
(BUF)	23013								
(EXB)	23016								
(LOB)	22735		00002	24026					
(RTN)	25571								
(FIL)	25560								
(IOH)	24030		00003	25745					
(ICD)	26037								
(TEF)	26036								
(RCH)	26035								
(ETT)	26034								
(REW)	26033								
(WEF)	26032								
(BSR)	26031								
(WRS)	26030								
(RDS)	26027								
(LUS)	25750								
(TRC)	26040								
(PRINT)	26333		00002	26074					
(EXEM)	26076								
(LIOU)	26707		00000	26704					
LOGICAL TAPE	MACHINE TAPE	TOTAL WRITES	TOTAL READS	NOISE RECORDS WRITING	TOTAL REDUNDANCIES WRITING	POSITIONING ERRORS			
1	A 1	0	493	0	0	0			
2	B 2	1008	1142	0	(II-1) 0	0			

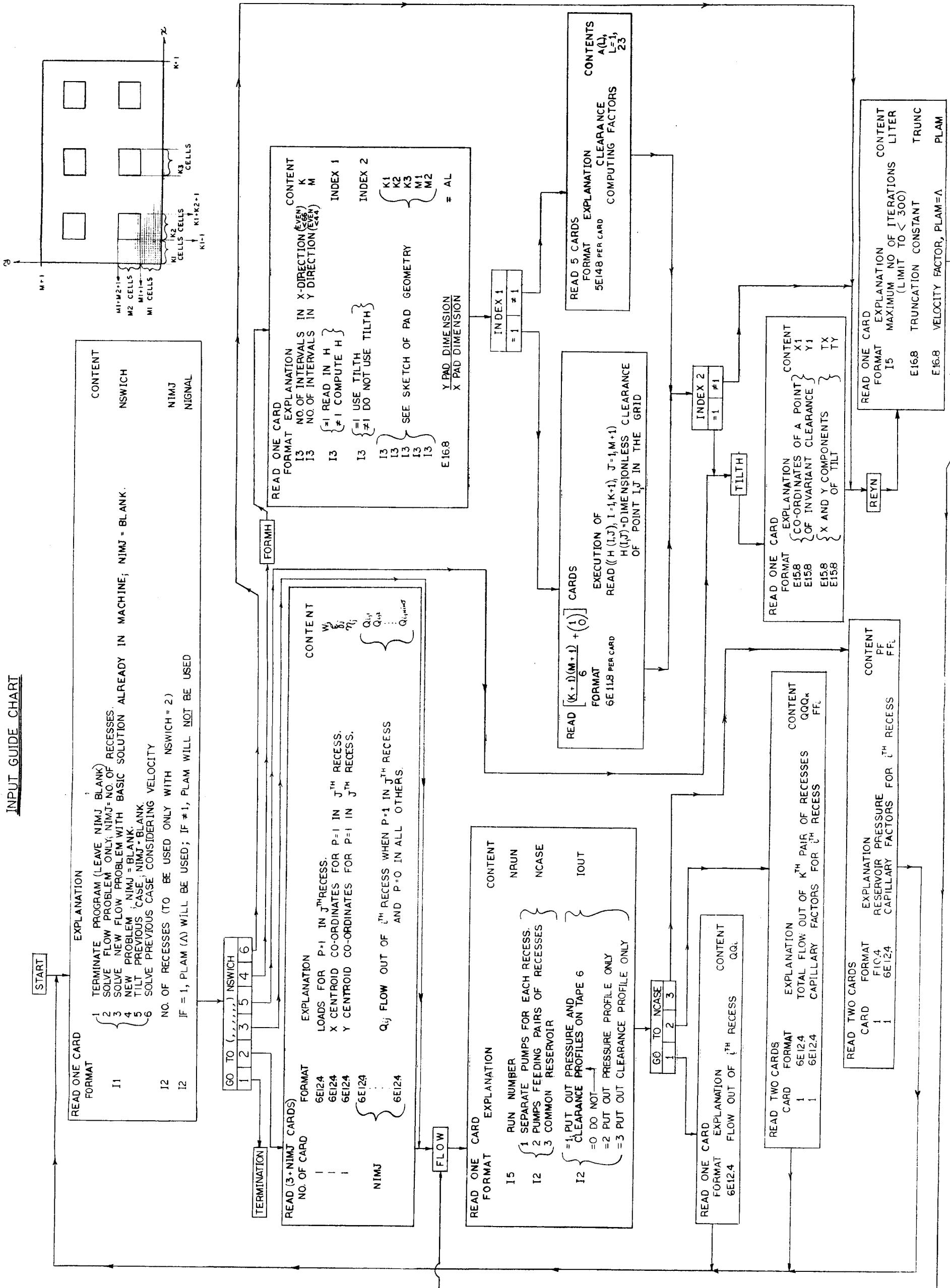
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APPENDIX III

INPUT GUIDE CHART

INPUT GUIDE CHART



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APPENDIX IV

SAMPLE PROBLEM INPUT DATA

AND

OUTPUT RESULTS

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HYDROSTATIC BEARING PROGRAM INPUT DATA

CROSS OUT CARDS NOT USED

1.	CARD	1      3      5	4      0	NSWICH	NIGNAL
2.	1 CARD	3      6      9      12      15      18      21      24      27      43	28      16      0      0      5      5      0      4      2	0.57142857	
		K      M      INDEX1      INDEX2      K1      K2      K3      M1      M2      AL	14      28      42      56		70
3.		1.0      A(1)	0.0      A(2)	0.0	0.0      0.0
4.		0.0	0.0	0.0	0.0      0.0
5.	5 CARDS (INCLUDE ALL 5)	0.0	0.0	0.0	0.0      0.0
6.		0.0	0.0	0.0	0.0      0.0
7.		0.0	0.0	0.0      A(23)	0.0      0.0
8.	+CARD	15	30	45	60
9.	1 CARD	5      21      37	1000      0.30      0.0	LITER      TRUNC      PLAM	
10.	1 CARD	5      7      9	7701      1      1	NRUN      NCASE      IOUT	
11.	1 CARD	12      24      36      48	1.0      1.0      1.0      1.0	—      —	60      72
12.	+CARD	10			
13.	+CARD	12      24      36      48			6      72
14.	+CARD	1			

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HYDROSTATIC BEARING PROGRAM INPUT DATA  
CROSS OUT CARDS NOT USED

1.	1 CARD	1 5 NSWICH	3 0 NIGNAL	5	3	9	12	15	18	21	24	27	43
2.	1 CARD												
3.				14		28		42		56		70	
4.													
5.	5 CARDS (INCLUDE ALL 5)												
6.													
7.													
8.	1 CARD	0.50 X1	0.28571429 Y1	1.0 TX			45		60				
9.	1 CARD	1000 LITER	0.30 TRUNC	1.0 PLAM		5 21 37							
10.	1 CARD	.7702 NRUN	1 NCASE	1 IOUT	5 7 9								
11.	1 CARD	1.0 QQ(1)	1.0 QQ(2)	1.0 QQ(3)	1.0 QQ(4)	12 24 36 48			60		72		
12.	1 CARD				10								
13.	1 CARD					12 24 36 48			60		72		
14.	1 CARD				1								

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HYDROSTATIC BEARING PROGRAM INPUT DATA F-B2099  
CROSS OUT CARDS NOT USED

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RUN 7701		PRESSURE DISTRIBUTION														PRESSURE DISTRIBUTION															
I=	J=	I= 1	I= 2	I= 3	I= 4	I= 5	I= 6	I= 7	I= 8	I= 9	I= 10	I= 11	I= 12	I= 13	I= 14	I= 15	I= 16	I= 17	I= 18	I= 19	I= 20										
1,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
2,	0.	0.	0.0117	0.0237	0.0362	0.0487	0.0596	0.0665	0.0699	0.0708	0.0708	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
3,	0.	0.	0.0230	0.0470	0.0725	0.0991	0.1232	0.1365	0.1425	0.1438	0.1438	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
4,	0.	0.	0.0334	0.0687	0.1077	0.1519	0.1977	0.2140	0.2198	0.2208	0.2208	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
5,	0.	0.	0.0419	0.0866	0.1379	0.2031	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
6,	0.	0.	0.0477	0.0980	0.1543	0.2209	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
7,	0.	0.	0.0508	0.1036	0.1604	0.2243	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
8,	0.	0.	0.0520	0.1051	0.1596	0.2143	0.2632	0.2835	0.2918	0.2942	0.2942	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
9,	0.	0.	0.0523	0.1052	0.1586	0.2101	0.2534	0.2770	0.2879	0.2912	0.2912	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
10,	0.	0.	0.0520	0.1051	0.1596	0.2143	0.2632	0.2835	0.2918	0.2942	0.2942	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
11,	0.	0.	0.0508	0.1036	0.1604	0.2243	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
12,	0.	0.	0.0477	0.0980	0.1543	0.2209	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
13,	0.	0.	0.0419	0.0866	0.1379	0.2031	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
14,	0.	0.	0.0334	0.0687	0.1077	0.1519	0.1977	0.2140	0.2198	0.2208	0.2208	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
15,	0.	0.	0.0230	0.0470	0.0725	0.0991	0.1232	0.1365	0.1425	0.1438	0.1438	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
16,	0.	0.	0.0117	0.0237	0.0362	0.0487	0.0596	0.0665	0.0699	0.0708	0.0708	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.										
		RUN 7701		PRESSURE DISTRIBUTION														PRESSURE DISTRIBUTION													
I=	J=	I= 11	I= 12	I= 13	I= 14	I= 15	I= 16	I= 17	I= 18	I= 19	I= 20	I= 11	I= 12	I= 13	I= 14	I= 15	I= 16	I= 17	I= 18	I= 19	I= 20										
1,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.				
2,	0.	0.	0.0657	0.1337	0.2082	0.3018	0.3018	0.3018	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
3,	0.	0.	0.0600	0.0995	0.1797	0.2340	0.2562	0.2618	0.2701	0.2711	0.2711	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
4,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
5,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
6,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
7,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
8,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
9,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
10,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
11,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
12,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
13,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
14,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
15,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
16,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					
17,	0.	0.	0.0545	0.1201	0.1797	0.2340	0.2649	0.2910	0.3018	0.3018	0.3018	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.					

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PRESSURE DISTRIBUTION													
RUN	7701	I= 21	I= 22	I= 23	I= 24	I= 25	I= 26	I= 27	I= 28	I= 29	I= 30	I= 31	I= 32
J= 1,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.
J= 2,	0.0708	0.0699	0.0665	0.0596	0.0487	0.0362	0.0237	0.0117	0.	0.	0.	0.	0.
J= 3,	0.1438	0.1425	0.1365	0.1232	0.0991	0.0725	0.0470	0.0230	0.	0.	0.	0.	0.
J= 4,	0.2208	0.2198	0.2140	0.1977	0.1519	0.1077	0.0687	0.0334	0.	0.	0.	0.	0.
J= 5,	0.3018	0.3018	0.3018	0.3018	0.2031	0.1379	0.0866	0.0419	0.	0.	0.	0.	0.
J= 6,	0.3018	0.3018	0.3018	0.3018	0.2209	0.1543	0.0980	0.0477	0.	0.	0.	0.	0.
J= 7,	0.3018	0.3018	0.3018	0.3018	0.2243	0.1604	0.1036	0.0508	0.	0.	0.	0.	0.
J= 8,	0.3942	0.2918	0.2835	0.2632	0.2143	0.1576	0.1051	0.0520	0.	0.	0.	0.	0.
J= 9,	0.2912	0.2879	0.2770	0.2534	0.2101	0.1586	0.1052	0.0523	0.	0.	0.	0.	0.
J= 10,	0.2942	0.2918	0.2835	0.2632	0.2143	0.1596	0.1051	0.0520	0.	0.	0.	0.	0.
J= 11,	0.3018	0.3018	0.3018	0.3018	0.2243	0.1604	0.1036	0.0508	0.	0.	0.	0.	0.
J= 12,	0.3018	0.3018	0.3018	0.3018	0.2209	0.1543	0.0980	0.0477	0.	0.	0.	0.	0.
J= 13,	0.3018	0.3018	0.3018	0.3018	0.2031	0.1379	0.0866	0.0419	0.	0.	0.	0.	0.
J= 14,	0.2208	0.2198	0.2140	0.1977	0.1519	0.1077	0.0687	0.0334	0.	0.	0.	0.	0.
J= 15,	0.1438	0.1425	0.1365	0.1232	0.0991	0.0675	0.0470	0.0230	0.	0.	0.	0.	0.
J= 16,	0.0708	0.0699	0.0665	0.0596	0.0487	0.0362	0.0237	0.0117	0.	0.	0.	0.	0.
J= 17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.

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RUN NUMBER 7702 HYDROSTATIC BEARING PAD WITH 4 RECESSES.  
 GRID POINTS, K = 28 M = 16 FEEDING = POSITIVE DISPLACEMENT PUMPS FEEDING EACH RECESS.  
 CLEARANCE COEFFICIENTS K1 = 5 K2 = 5 K3 = 0 K4 = 4 M1 = 4 M2 = 2  
 AA(1) = 0.09999999E 01 ,AA  
 RECESS PRESSURES AL(1) = 0.60023 , AL(2) = 0.17013 , AL(3) = 0.17013 , AL(4) = 0.60023 , ALL

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RUN 7702		PRESSURE DISTRIBUTION													
I=	J=	I= 1	I= 2	I= 3	I= 4	I= 5	I= 6	I= 7	I= 8	I= 9	I= 10				
1,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.				
2,	0.	0.0337	0.0623	0.0878	0.1100	0.1273	0.1367	0.1398	0.1379	0.	0.1310				
3,	0.	0.0661	0.1230	0.1746	0.2218	0.2603	0.2788	0.2846	0.2816	0.	0.2694				
4,	0.	0.0956	0.1788	0.2572	0.3353	0.4088	0.4321	0.4342	0.4356	0.4236	0.				
5,	0.	0.1196	0.2245	0.3265	0.4408	0.6002	0.6002	0.6002	0.6002	0.6002	0.6002				
6,	0.	0.1360	0.2539	0.3649	0.4784	0.6002	0.6002	0.6002	0.6002	0.6002	0.6002				
7,	0.	0.1453	0.2691	0.3810	0.4883	0.6002	0.6002	0.6002	0.6002	0.6002	0.6002				
8,	0.	0.1494	0.2746	0.3827	0.4757	0.5482	0.5745	0.5819	0.5782	0.5627	0.				
9,	0.	0.1505	0.2757	0.3820	0.4701	0.5367	0.5653	0.5696	0.5696	0.5495	0.				
10,	0.	0.1494	0.2746	0.3827	0.4757	0.5482	0.5745	0.5819	0.5782	0.5627	0.				
11,	0.	0.1453	0.2691	0.3810	0.4883	0.6002	0.6002	0.6002	0.6002	0.6002	0.6002				
12,	0.	0.1360	0.2539	0.3649	0.4784	0.6002	0.6002	0.6002	0.6002	0.6002	0.6002				
13,	0.	0.1196	0.2245	0.3265	0.4408	0.6002	0.6002	0.6002	0.6002	0.6002	0.6002				
14,	0.	0.0956	0.1788	0.2572	0.3353	0.4088	0.4321	0.4342	0.4356	0.4236	0.				
15,	0.	0.0661	0.1300	0.1746	0.2218	0.2603	0.2788	0.2846	0.2816	0.2694	0.				
16,	0.	0.0337	0.0623	0.0878	0.1100	0.1273	0.1367	0.1398	0.1379	0.	0.1310				
17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.				

RUN 7702		PRESSURE DISTRIBUTION													
I=	J=	I= 11	I= 12	I= 13	I= 14	I= 15	I= 16	I= 17	I= 18	I= 19	I= 20				
1,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.				
2,	0.	0.0997	0.0911	0.0658	0.0546	0.0473	0.0433	0.0416	0.0413	0.	0.				
3,	0.	0.2443	0.2020	0.1615	0.1297	0.1071	0.0926	0.0849	0.0823	0.0828	0.0830				
4,	0.	0.3922	0.3084	0.2387	0.1885	0.1546	0.1335	0.1228	0.1206	0.1248	0.1258				
5,	0.	0.6002	0.4111	0.2377	0.1942	0.1674	0.1540	0.1532	0.1701	0.1701	0.1701				
6,	0.	0.6002	0.4601	0.3439	0.2716	0.2231	0.1921	0.1747	0.1683	0.1701	0.1701				
7,	0.	0.6002	0.4601	0.3220	0.2914	0.2416	0.2081	0.1875	0.1763	0.1701	0.1701				
8,	0.	0.5267	0.4444	0.3650	0.3001	0.2512	0.2170	0.1947	0.1813	0.1739	0.1705				
9,	0.	0.5081	0.4377	0.3648	0.3024	0.2542	0.2197	0.1969	0.1830	0.1750	0.1705				
10,	0.	0.5267	0.4444	0.3650	0.3001	0.2512	0.2170	0.1947	0.1813	0.1739	0.1705				
11,	0.	0.6002	0.4601	0.3220	0.2914	0.2416	0.2081	0.1875	0.1763	0.1701	0.1701				
12,	0.	0.6002	0.4496	0.3439	0.2716	0.2231	0.1921	0.1747	0.1683	0.1701	0.1701				
13,	0.	0.6002	0.4111	0.3048	0.2377	0.1942	0.1674	0.1540	0.1532	0.1701	0.1701				
14,	0.	0.3922	0.3084	0.2387	0.1885	0.1546	0.1335	0.1228	0.1206	0.1248	0.1258				
15,	0.	0.2443	0.2020	0.1615	0.1297	0.1071	0.0926	0.0849	0.0823	0.0828	0.0830				
16,	0.	0.1182	0.0997	0.0911	0.0658	0.0546	0.0473	0.0433	0.0416	0.	0.				
17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.				

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CLEARANCE DISTRIBUTION

RUN	7702	I= 1	I= 2	I= 3	I= 4	I= 5	I= 6	I= 7	I= 8	I= 9	I= 10
J= 1,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 2,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 3,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 4,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 5,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 6,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 7,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 8,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 9,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 10,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 11,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 12,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 13,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 14,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 15,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 16,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	
J= 17,	0.5000000	0.5357143	0.5714286	0.6071429	0.6428571	0.6785714	0.7142857	0.7500000	0.7857143	0.8214286	

CLEARANCE DISTRIBUTION

RUN	7702	I= 11	I= 12	I= 13	I= 14	I= 15	I= 16	I= 17	I= 18	I= 19	I= 20
J= 1,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 2,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 3,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 4,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 5,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 6,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 7,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 8,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 9,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 10,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 11,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 12,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 13,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 14,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 15,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 16,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	
J= 17,	0.8928571	0.8928571	0.9285714	0.9642857	1.0000000	1.0357143	1.0714286	1.1071429	1.1428571	1.1785714	

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RUN	7702	I = 21	I = 22	I = 23	I = 24	I = 25	I = 26	I = 27	I = 28	I = 29
J= 1,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 2,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 3,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 4,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 5,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 6,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 7,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 8,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 9,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 10,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 11,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 12,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 13,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 14,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 15,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 16,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	
J= 17,	1.2142857	1.2500000	1.2857143	1.3214286	1.3571429	1.3928571	1.4285714	1.4642857	1.5000000	

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PRESSURE DISTRIBUTION

RUN	7703	I= 1	I= 2	I= 3	I= 4	I= 5	I= 6	I= 7	I= 8	I= 9	I= 10
J= 1,	0.	0.	0.	0.	0.0672	0.0946	0.1185	0.1371	0.1473	0.1507	0.1486
J= 2,	0.	0.	0.	0.	0.1325	0.1881	0.2390	0.2805	0.3067	0.3034	0.2903
J= 3,	0.	0.	0.	0.	0.1927	0.2771	0.3613	0.4406	0.4656	0.4722	0.4694
J= 4,	0.	0.	0.	0.	0.2419	0.3519	0.4750	0.6468	0.6468	0.6468	0.4565
J= 5,	0.	0.	0.	0.	0.2736	0.3932	0.5155	0.6468	0.6468	0.6468	0.6468
J= 6,	0.	0.	0.	0.	0.2900	0.4106	0.5262	0.6468	0.6468	0.6468	0.6468
J= 7,	0.	0.	0.	0.	0.1566	0.2959	0.4123	0.5126	0.5907	0.6270	0.6063
J= 8,	0.	0.	0.	0.	0.1610	0.2959	0.4116	0.5066	0.5762	0.6091	0.6138
J= 9,	0.	0.	0.	0.	0.1621	0.2971	0.4123	0.5126	0.5907	0.6192	0.5922
J= 10,	0.	0.	0.	0.	0.1610	0.2959	0.4123	0.5126	0.6190	0.6230	0.6063
J= 11,	0.	0.	0.	0.	0.1566	0.2900	0.4106	0.5262	0.6468	0.6468	0.6468
J= 12,	0.	0.	0.	0.	0.1466	0.2736	0.3932	0.5155	0.6468	0.6468	0.6468
J= 13,	0.	0.	0.	0.	0.1289	0.2419	0.3519	0.4750	0.6468	0.6468	0.6468
J= 14,	0.	0.	0.	0.	0.1030	0.1927	0.2771	0.3613	0.4406	0.4656	0.4694
J= 15,	0.	0.	0.	0.	0.0712	0.1325	0.1881	0.2390	0.2805	0.3005	0.3034
J= 16,	0.	0.	0.	0.	0.0363	0.0672	0.0946	0.1185	0.1371	0.1507	0.1486
J= 17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.

PRESSURE DISTRIBUTION

RUN	7703	I= 11	I= 12	I= 13	I= 14	I= 15	I= 16	I= 17	I= 18	I= 19	I= 20
J= 1,	0.	0.	0.	0.	0.0874	0.0710	0.0589	0.0511	0.0467	0.0449	0.0445
J= 2,	0.	0.	0.	0.	0.1273	0.2177	0.1741	0.1398	0.1155	0.0999	0.0894
J= 3,	0.	0.	0.	0.	0.2632	0.4226	0.324	0.2573	0.2033	0.1325	0.1348
J= 4,	0.	0.	0.	0.	0.4430	0.6468	0.386	0.2562	0.2094	0.1806	0.1655
J= 5,	0.	0.	0.	0.	0.4846	0.6468	0.3707	0.2929	0.2406	0.2073	0.1886
J= 6,	0.	0.	0.	0.	0.4959	0.6468	0.3902	0.3141	0.2605	0.2246	0.2024
J= 7,	0.	0.	0.	0.	0.5676	0.4790	0.3935	0.3235	0.2710	0.2341	0.2101
J= 8,	0.	0.	0.	0.	0.5475	0.5475	0.3933	0.2260	0.2741	0.2371	0.2126
J= 9,	0.	0.	0.	0.	0.5676	0.4790	0.3935	0.3235	0.2710	0.2341	0.2101
J= 10,	0.	0.	0.	0.	0.6468	0.4959	0.3902	0.3141	0.2605	0.2246	0.2024
J= 11,	0.	0.	0.	0.	0.6468	0.4846	0.3707	0.2929	0.2406	0.2073	0.1886
J= 12,	0.	0.	0.	0.	0.6468	0.4430	0.3286	0.3141	0.2605	0.2246	0.2024
J= 13,	0.	0.	0.	0.	0.6468	0.4430	0.3286	0.3141	0.2605	0.2246	0.2024
J= 14,	0.	0.	0.	0.	0.4226	0.3324	0.2573	0.2033	0.1668	0.1441	0.1217
J= 15,	0.	0.	0.	0.	0.2632	0.2177	0.1741	0.1398	0.1155	0.0999	0.0894
J= 16,	0.	0.	0.	0.	0.1074	0.0874	0.0710	0.0589	0.0511	0.0467	0.0445
J= 17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.

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RUN	7703	I=	21	I=	22	I=	23	I=	24	I=	25	I=	26	I=	27	I=	28	I=	29
J=	1,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.
J=	2,	0.0440	0.0425	0.0440	0.0425	0.0397	0.0349	0.0349	0.0277	0.0277	0.0277	0.0199	0.0199	0.0126	0.0126	0.0060	0.0060	0.	0.
J=	3,	0.0889	0.0867	0.0889	0.0867	0.0819	0.0725	0.0725	0.0566	0.0566	0.0566	0.0401	0.0401	0.0250	0.0250	0.0118	0.0118	0.	0.
J=	4,	0.1356	0.1337	0.1356	0.1337	0.1291	0.1178	0.1178	0.0875	0.0875	0.0875	0.0598	0.0598	0.0367	0.0367	0.0172	0.0172	0.	0.
J=	5,	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1182	0.1182	0.1182	0.0770	0.0770	0.0464	0.0464	0.0216	0.0216	0.	0.
J=	6,	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1286	0.1286	0.1286	0.0861	0.0861	0.0526	0.0526	0.0246	0.0246	0.	0.
J=	7,	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1303	0.1303	0.1303	0.0893	0.0893	0.0554	0.0554	0.0262	0.0262	0.	0.
J=	8,	0.1814	0.1777	0.1804	0.1777	0.1708	0.1562	0.1562	0.1231	0.1231	0.1231	0.0884	0.0884	0.0561	0.0561	0.0298	0.0298	0.	0.
J=	9,	0.1804	0.1773	0.1814	0.1773	0.1662	0.1492	0.1492	0.1201	0.1201	0.1201	0.0876	0.0876	0.0561	0.0561	0.0269	0.0269	0.	0.
J=	10,	0.1814	0.1777	0.1814	0.1777	0.1708	0.1562	0.1562	0.1231	0.1231	0.1231	0.0884	0.0884	0.0561	0.0561	0.0268	0.0268	0.	0.
J=	11,	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1303	0.1303	0.1303	0.0893	0.0893	0.0554	0.0554	0.0262	0.0262	0.	0.
J=	12,	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1286	0.1286	0.1286	0.0861	0.0861	0.0526	0.0526	0.0246	0.0246	0.	0.
J=	13,	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1838	0.1182	0.1182	0.1182	0.0770	0.0770	0.0464	0.0464	0.0216	0.0216	0.	0.
J=	14,	0.1356	0.1337	0.1356	0.1337	0.1291	0.1178	0.1178	0.0875	0.0875	0.0875	0.0598	0.0598	0.0367	0.0367	0.0172	0.0172	0.	0.
J=	15,	0.0889	0.0867	0.0889	0.0867	0.0819	0.0725	0.0725	0.0566	0.0566	0.0566	0.0401	0.0401	0.0250	0.0250	0.0118	0.0118	0.	0.
J=	16,	0.0440	0.0425	0.0440	0.0425	0.0397	0.0349	0.0349	0.0277	0.0277	0.0277	0.0199	0.0199	0.0126	0.0126	0.0060	0.0060	0.	0.
J=	17,	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.	0.